

DESIGN OF MACHINE ELEMENTS-I

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COURSE OBJECTIVES

- To introduce the concepts of design of machine elements.
- To familiarize with fundamental approaches to failure prevention for static and dynamic loading.
- To explain the design procedures of different types of joints.
- To familiarize with various theories related the design of machine elements for different loading conditions.

SYLLABUS

Unit 1 Design for Static Loads

Introduction: General considerations of design, design process. Selection of Engineering materials - properties – Manufacturing considerations in the design. BIS codes of materials. Preferred numbers. Design for static loads: Modes of failure, Design of components subjected to axial, bending, torsion and combined loads. Theories of failure for static loads.

Unit 2 Design for Dynamic Loads

Stress concentration –notch sensitivity, Types of fluctuating loads– Design for fluctuating stresses –finite life - Endurance limit - Estimation of Endurance limit – Soderberg, Goodman and Modified Goodman criterion for fatigue failure.

Unit 3 Design of Bolted and Welded joints

Design of Bolted Joints: Preload of bolts, various stresses induced in the bolts, torque requirement for bolt tightening, eccentrically loaded joints.

Design of Welded Joints: Stresses of lap and butt welds, eccentrically loaded welded joints. Joints subjected to bending and torsion.

Unit 4 Design of Keys, Cotter and Knuckle Joints

Keys: Design of sunk, saddle, Kennedy and woodruff keys.

Cotter and Knuckle Joints: Socket and spigot joint, Sleeve and Cotter joints, Gib and Cotter joint, Knuckle joint.

Unit 5 Design of Shafts and Couplings

Shafts: Design of solid and hollow shafts bending, torsion, axial and combined bending and axial loading.

Shaft Couplings: Design of Rigid couplings-Muff, Split muff and Flange couplings-Flexible couplings: bushed pin type.

COURSE OUTCOMES

A student will be able to

1. Summarize the basic concepts of design of machine elements
2. Analyse the components subjected to dynamic loading
3. Design bolted and welded joints subjected to given loads.
4. Design keys, cotter and knuckle joints for various applications.
5. Design shafts and shaft couplings for given conditions.

OUTLINE

- Introduction
- Definition
- General considerations of design
- Design process
- Selection of Engineering materials
- properties
- Manufacturing considerations in the design.
- BIS codes of materials.
- Preferred numbers

Unit 1 Design for Static Loads

Introduction: General considerations of design, design process. Selection of Engineering materials - properties – Manufacturing considerations in the design. BIS codes of materials. Preferred numbers. Design for static loads: Modes of failure, Design of components subjected to axial, bending, torsion and combined loads. Theories of failure for static loads.

Course Outcome 1

At the end of the unit, student will be able to Summarize the basic concepts of design of machine elements

Definition of Design

- Design is to formulate a plan to satisfy a particular need and to create something with physical reality.
- Realization of a concept or idea into a configuration
- Design is the creation of a plan or convention for the construction of an object, system or measurable human interaction.
- Design is nothing but Decision Making

Engineering Design:

- Engineering design is an iteration process of devising a system to meet a set of desired needs. The design process begins with an identified need, which can be initiated based on any of the following sources:
 - Observable deficiency (i.e., A car bumper that gets easily damaged in low speed collisions)
 - Product improvement (i.e., A system for improving traction on ice without studs or chains)
 - New products (i.e., A more effective alarm clock for reluctant students)
 - Change in law (i.e., An average automobile gas mileage of 25 miles per gallon of gasoline)

Machine Design

- Machine is a combination of several machine elements arranged to work together as a whole to accomplish specific purpose.
- Machine design involves designing the elements and arranging them optimally to obtain some useful work.
- Machine design is defined as “**the use of scientific principles, technical information and imagination in the description of a machine or a mechanical system to perform specific functions with maximum economy and efficiency**”.

General Considerations of Design

- Type of load and stresses caused by the load
- Motion of parts or kinematics of the Machine
- Selection of Materials
- Form and size of the parts
- Frictional resistance and lubrication
- Convenient and Economical features
- Use of standard parts
- Safety of Operation
- Workshop facilities
- Number of Machines to be manufactured
- Cost of Construction
- Assembling

DESIGN CONSIDERATIONS

- Strength
- Rigidity
- Reliability
- Safety
- Cost
- Weight
- Ergonomics
- Aesthetics
- Manufacturing considerations
- Assembly considerations
- Conformance to standards
- Friction and wear
- Life
- Vibrations
- Thermal considerations
- Lubrication
- Maintenance
- Flexibility
- Size and shape
- Stiffness
- Corrosion
- Noise
- Environmental considerations

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element

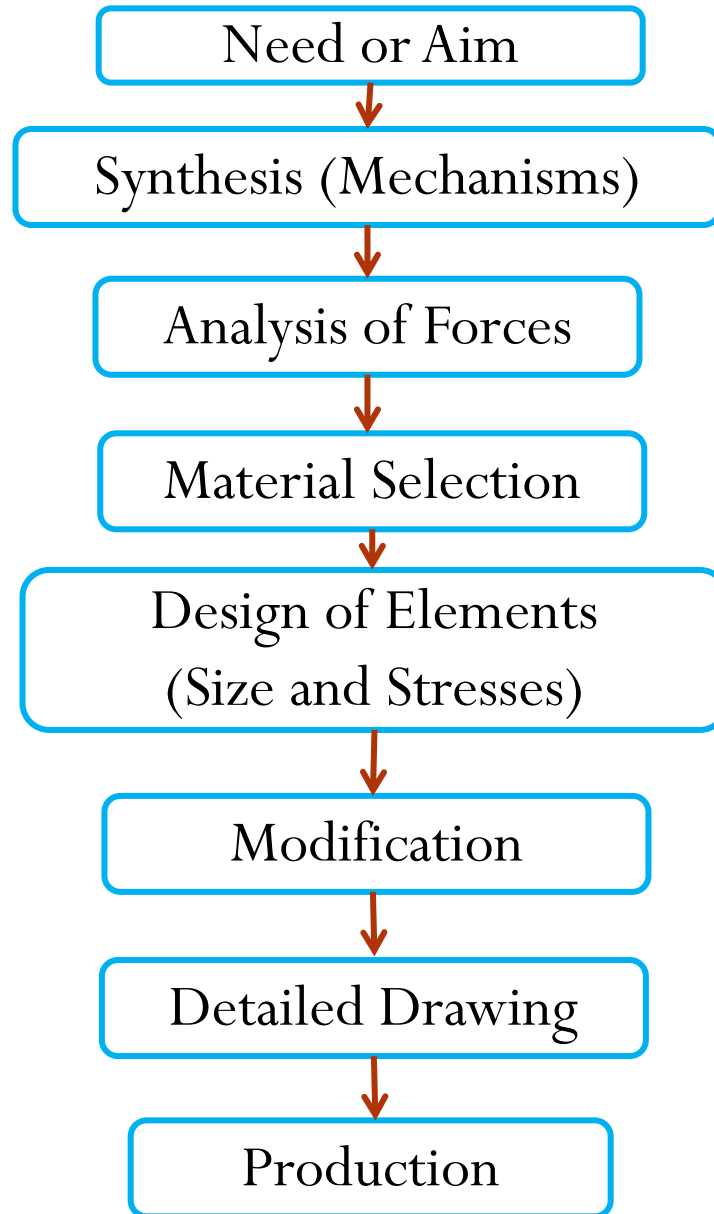


Modify Dimensions for Assembly and
Manufacture and
Check Design at Critical Cross-sections



Prepare Working Drawing of Element

General Procedure in Machine Design



Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 1: Specification of Functions of element

The design of Machine elements begins with the specification of the functions of element.

Functions of Element are:

1. Bearing
2. Key
3. Spring in Clock
4. Spring in Spring balance
5. Screw Fastening
6. Power Screw

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 2: Determination of Forces

In many cases a free-body diagram of forces is constructed to determine the forces acting different parts of the machine.

Forces are:

1. Force due to **Frictional Resistance**
2. Inertia Force due to **change in linear or angular velocity**
3. Centrifugal force due to **change in direction of velocity**
4. Static force due to **deadweight of machine part**
5. External force due to **energy, power transmission**
6. Force due to **thermal gradient or variation in temperature**

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 3: Selection of Material

Four basic factors which are considered in selecting the material are:

1. Availability
2. Cost
3. Mechanical Properties
4. Manufacturing considerations

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 4: Failure Criterion

Before finding out the dimensions of the component, it is necessary to know the type of failure that the component may fail when put into service

The three basic types of failure are as follows:

- 1 Failure by elastic deformation
- 2 Failure by general yielding
- 3 Failure by fracture

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 5: Determination of Dimensions

The geometric dimensions of the component are determined on the basis of failure criterion.

In simple cases, the dimensions are determined on the basis of allowable stress or deflection .

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 6: Design Modifications

The geometric dimensions of the component are modified from assembly and manufacturing considerations.

Basic Procedure of Design of Machine elements

Specify Functions of Element



Determine the forces acting on Element



Select Suitable Material for Element



Determine Failure Mode of Element



Determine Geometric Dimensions of element



Modify Dimensions for Assembly and Manufacture and Check Design at Critical Cross-sections



Prepare Working Drawing of Element

Step 7: Working drawing

The last step in the design of machine elements is to prepare a working drawing of the machine element showing:

1. Dimensions
2. Tolerances
3. Surface finish grades
4. Geometric tolerances
5. Special Production requirements like heat treatment

Selection of Engineering Materials

- a) Metals
- b) Non-metals

Metals:

Ferrous – Which contains iron as the major constituent

Ex. Steel, Cast Iron

Non-ferrous – materials don't contains Iron.

Ex. Copper, Aluminium

Non-Metals:

- (i) Ceramic materials – oxides, carbides and nitrides of various metals. Ex. Glass, Brick, Concrete, Cement etc.
- (ii) Organic materials – Polymeric materials composed of carbon compounds. Ex: Paper, fuel, rubber, paints, etc.

Factors to be considered for the selection of materials

1. Availability
2. Cost
3. Physical properties
4. Mechanical Properties
5. Manufacturing process

Physical properties:

- Color
 - Shape
 - Size
 - Density
- Electrical conductivity
- Thermal conductivity

Manufacturing Processes

- Primary Process
- Machining Process
- Surface Finishing Process
- Joining Process
- Process effecting change in properties like heat treatment, cold & hot working

Manufacturing Considerations in Design

- Design considerations in Casting
- Design considerations in Machining
- Design considerations in Forging

Design Considerations in Casting

- Design the part so that the shape is cast easily
- Select a casting process and material suitable for the part, size, mechanical properties, etc.
- Locate the parting line of the mold in the part.
- Locate and design the gates to allow uniform feeding of the mold cavity with molten material.
- Select an appropriate runner geometry for the system.
- Locate mold features such as sprue, screens and risers, as appropriate
- Make sure proper controls and good practices are in place.

Contd...

- Avoid corners, Angles and section thickness

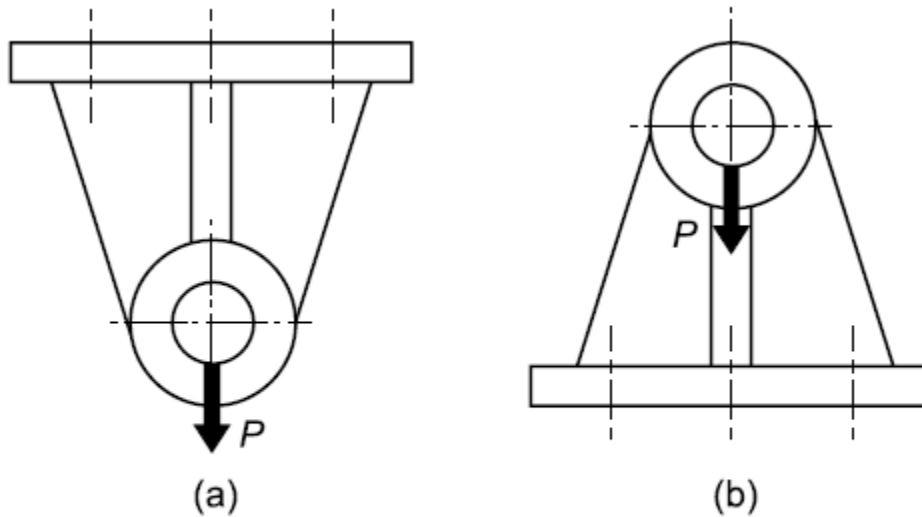


Fig. 3.1 (a) *Incorrect (Part in Tension)* (b) *Correct (Part in Compression)*

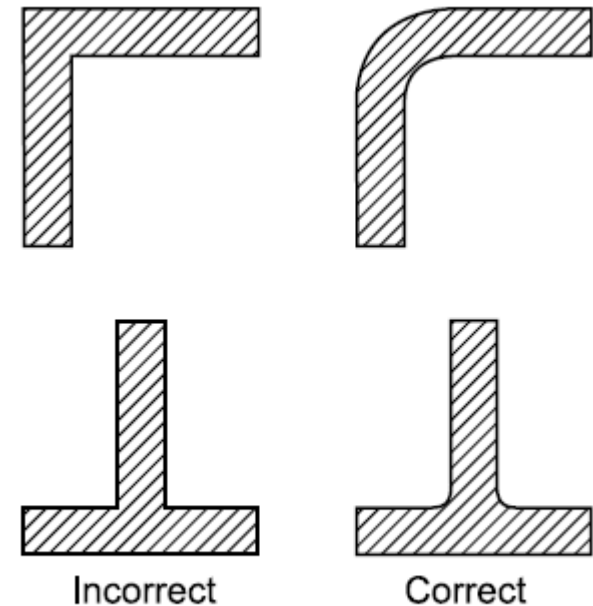


Fig. 3.3 *Provision of Fillet Radius*

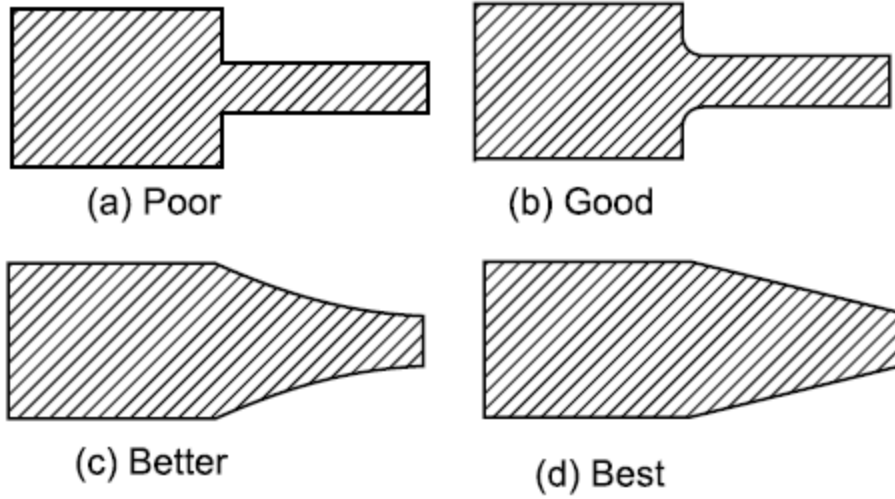


Fig. 3.4 *Change in Section-thickness*

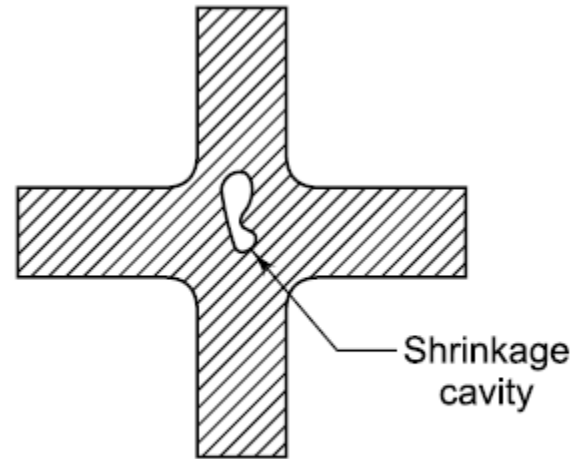


Fig. 3.5

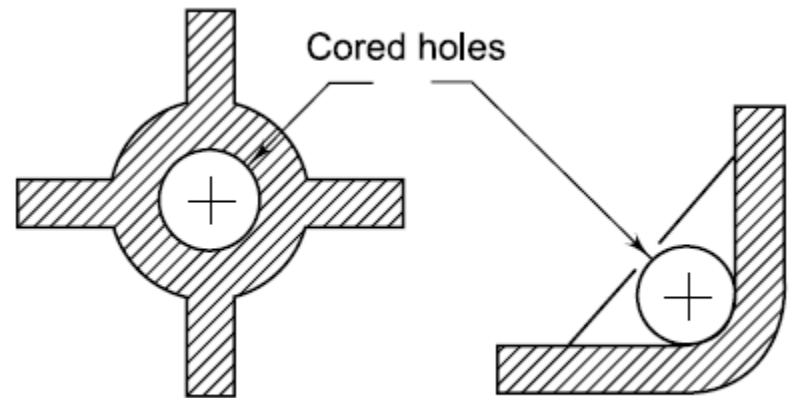


Fig. 3.6 *Cored Holes*

Design Considerations for Forging

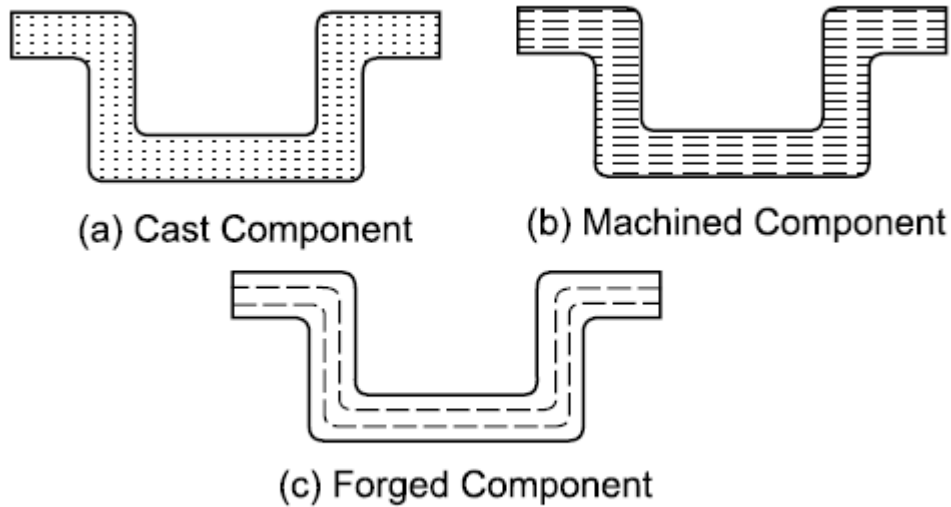


Fig. 3.12 *Grain Structure*

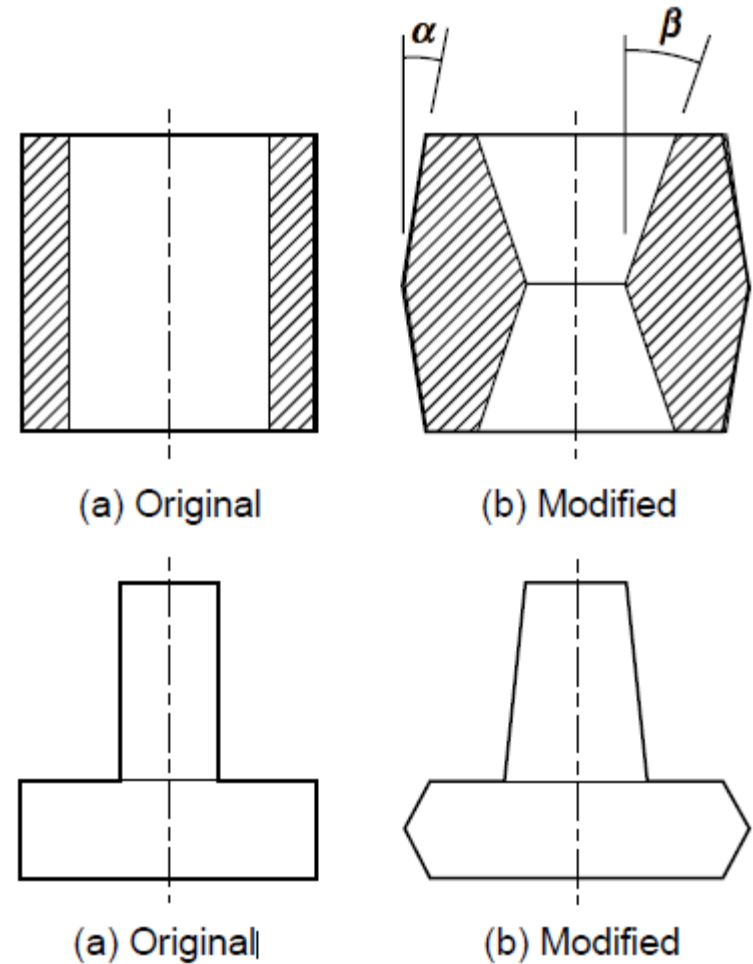
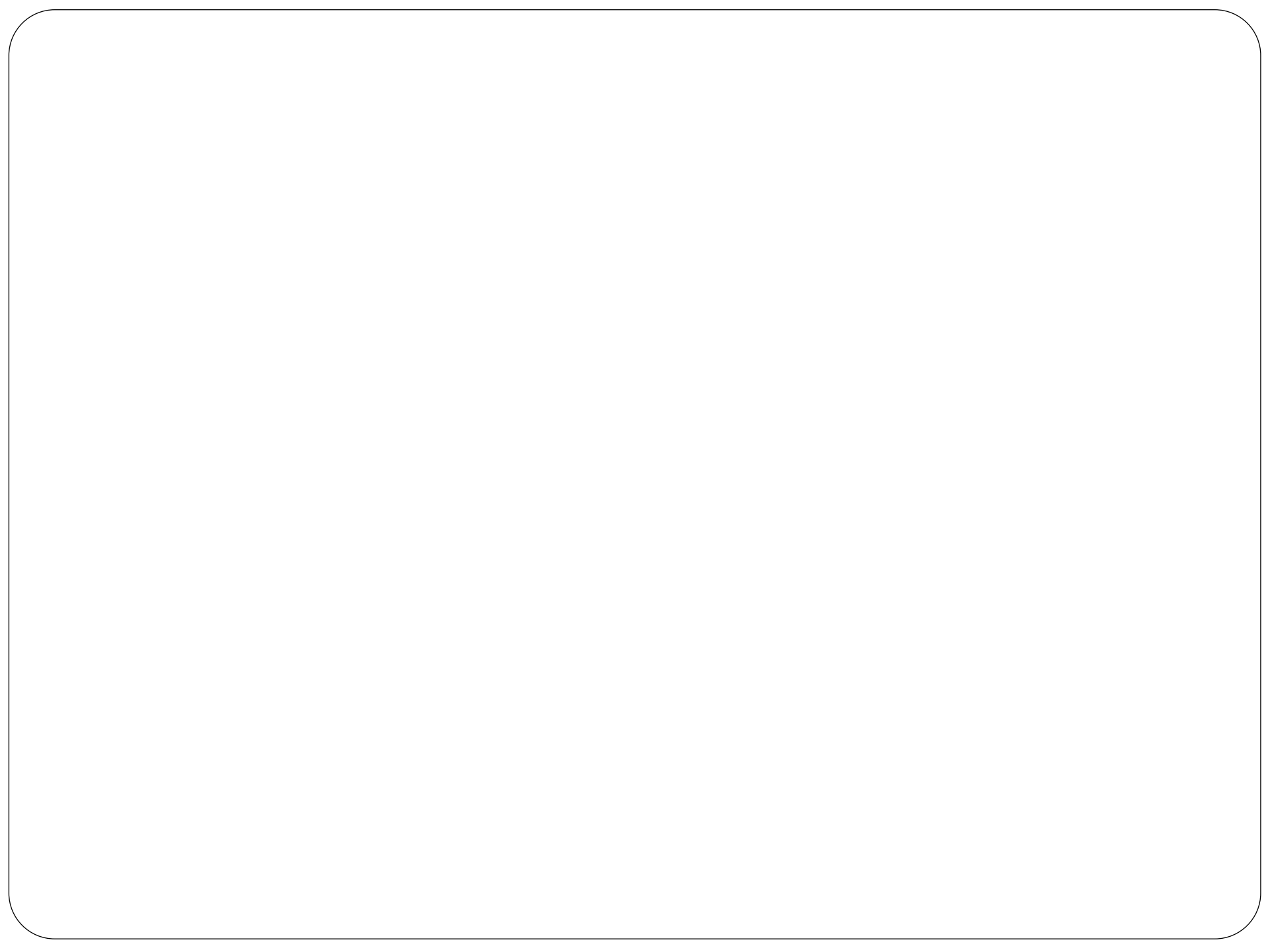
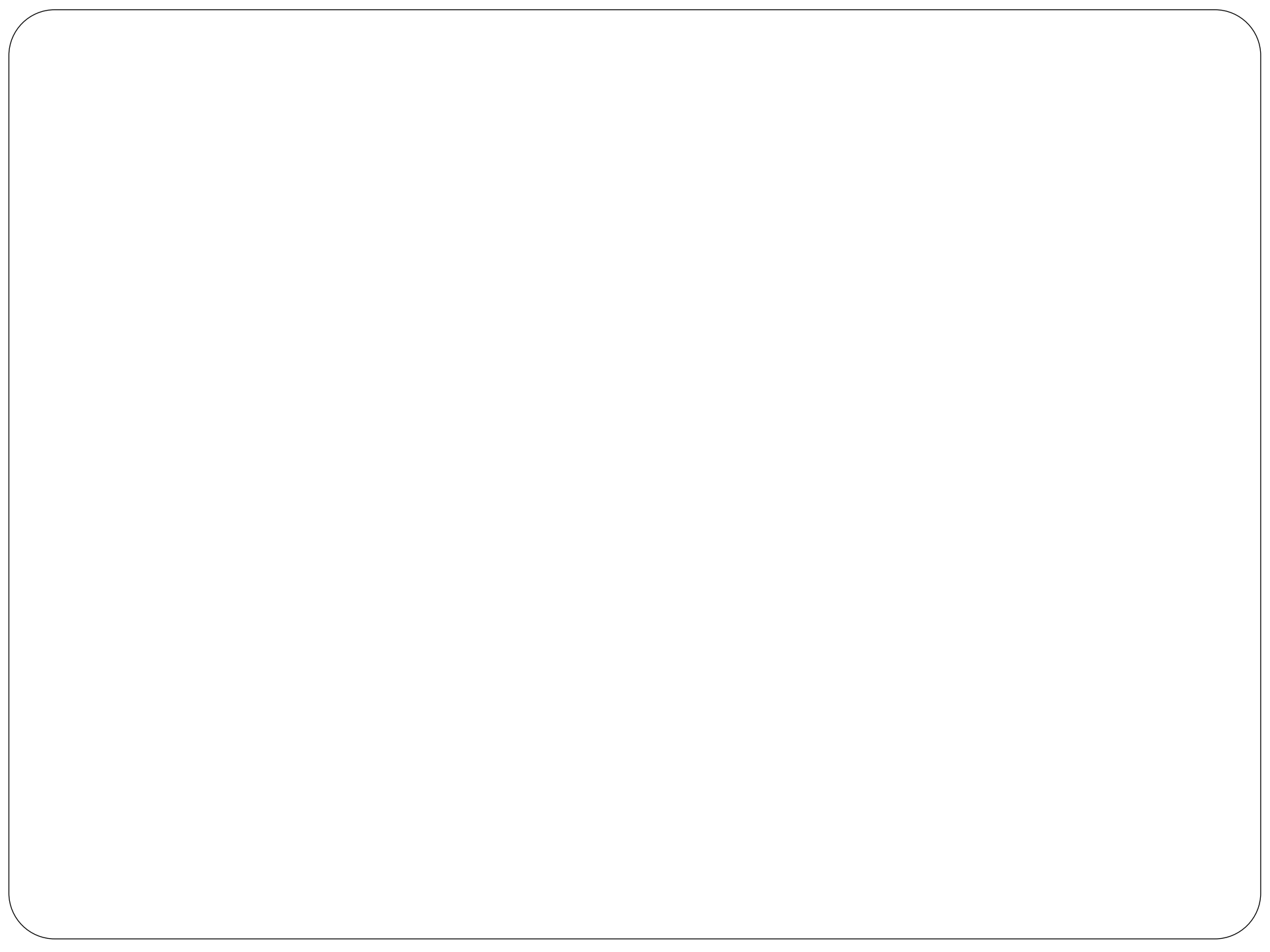


Fig. 3.13 *Draft for Forgings*

Design Considerations of Machining

- Avoid Machining
- Specify Liberal Tolerances
- Avoid sharp corners
- Use stock dimensions
- Design rigid parts
- Avoid shoulders and undercuts
- Avoid hard Materials





Stress Concentration

Presented by

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Department of Mechanical Engineering

Unit 2 Design for Dynamic Loads

Stress concentration –notch sensitivity, Types of fluctuating loads– Design for fluctuating stresses – finite life - Endurance limit - Estimation of Endurance limit – Soderberg, Goodman and Modified Goodman criterion for fatigue failure.

Course Outcome 2

At the end of the unit, student will be able to Analyse the components subjected to dynamic loading

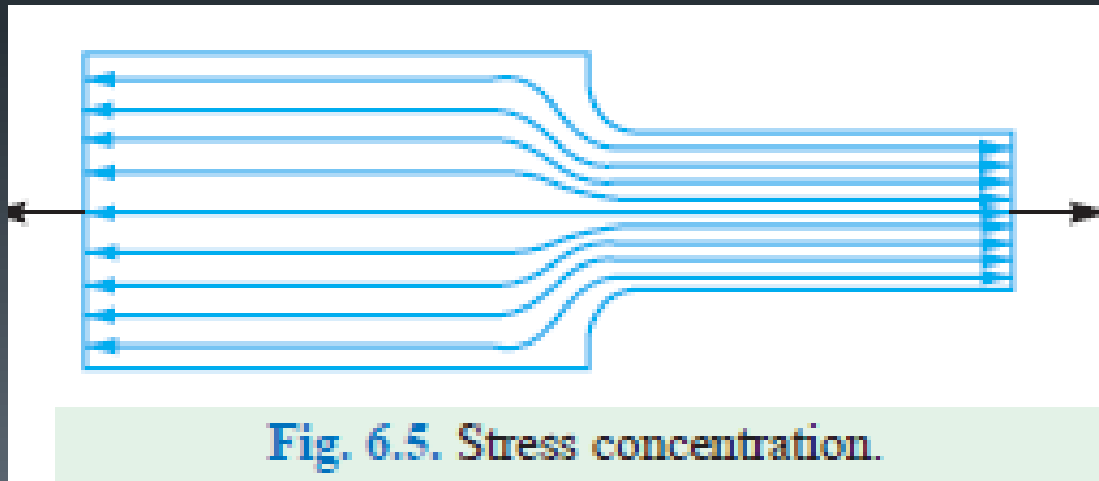
Introduction



1. Stress Concentration, Notch Sensitivity
2. Fluctuating stresses
3. Design for fluctuating stresses
4. Finite life
5. Endurance Limit
6. Soderberg, Goodman lines & Modified Goodman criterion for fatigue failure.

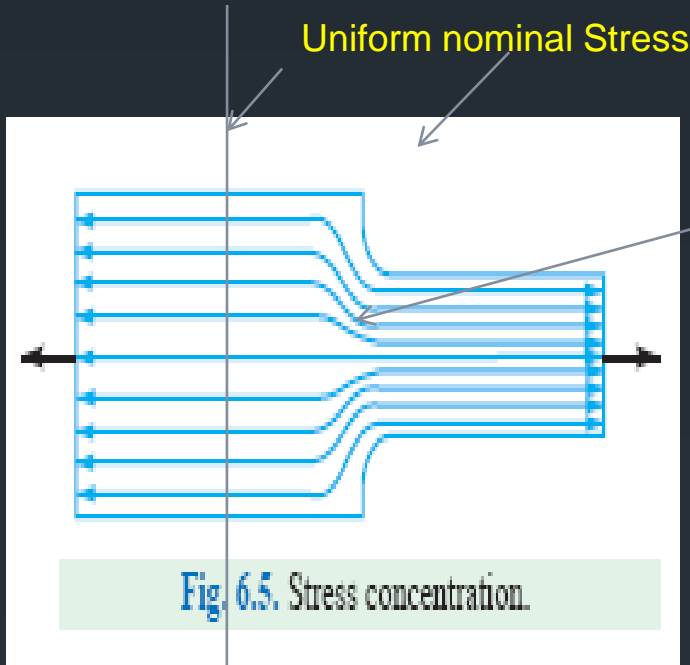
Stress Concentration

- Whenever a machine component changes the shape of its cross-section, the simple stress distribution no longer holds good and the neighbourhood of the discontinuity is different. This irregularity in the stress distribution caused by abrupt changes of form is called stress concentration.
- It occurs for all kinds of stresses in the presence of fillets, notches, holes, keyways, splines, surface roughness or scratches etc.



Concept of Stress Concentration

Stress Concentration



K_t is used for normal stresses and K_{ts} for shear stresses.

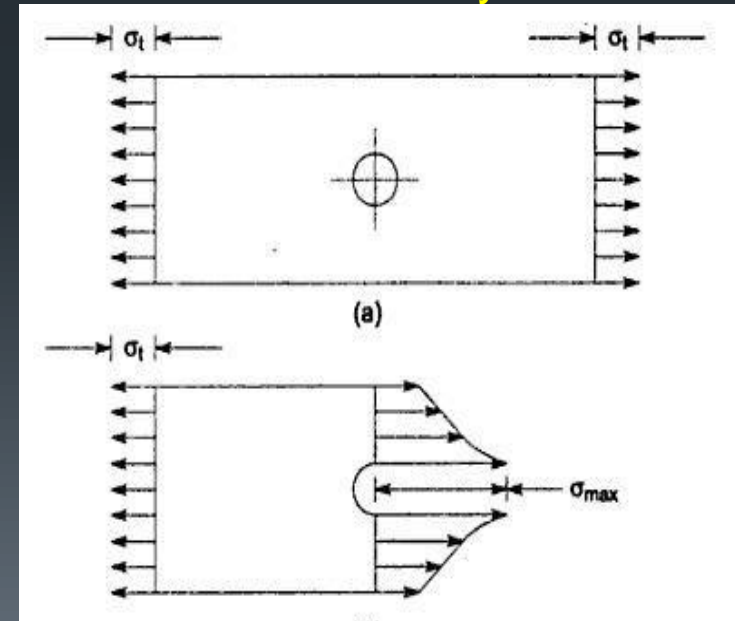
$$K = \frac{\sigma_{\max}}{\sigma_0}$$

$$\sigma_0 = \frac{P}{(W - d)t}$$

Theoretical stress concentration factor, K_t

$$K_t = \frac{\sigma_{\max}}{\sigma_0} \quad K_{ts} = \frac{\tau_{\max}}{\tau_0}$$

Nominal stress, max stress with no discontinuity



Stress Concentration :

- It is defined as the localization of high stress due to irregularities present in the component and abrupt changes of the cross section

- $$K_t = \frac{\text{Highest value of actual stress near discontinuity}}{\text{normal stress obtained by elementary equations for minimum cross-section}}$$

or

- $$K_t = \frac{\text{Maximum stress}}{\text{nominal stress}}$$

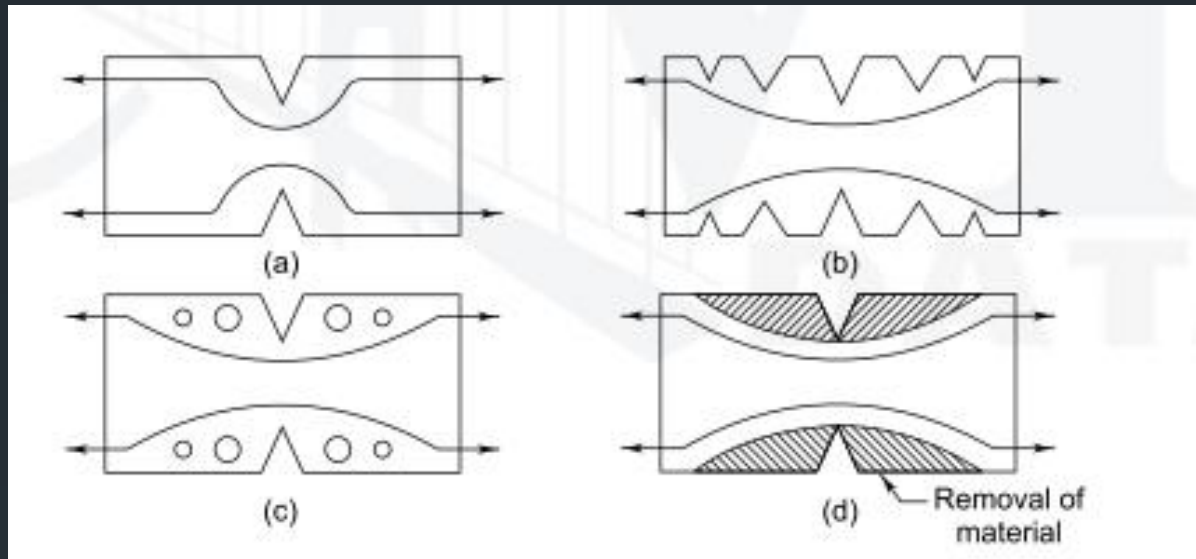
Causes

- Abrupt Change of c/s
- Poor surface finish
- Localized loading
- Variation in the material properties

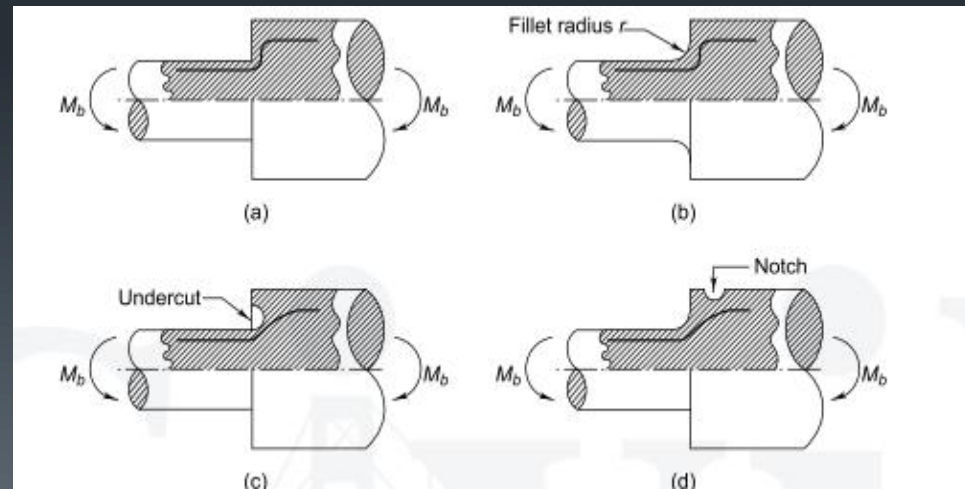
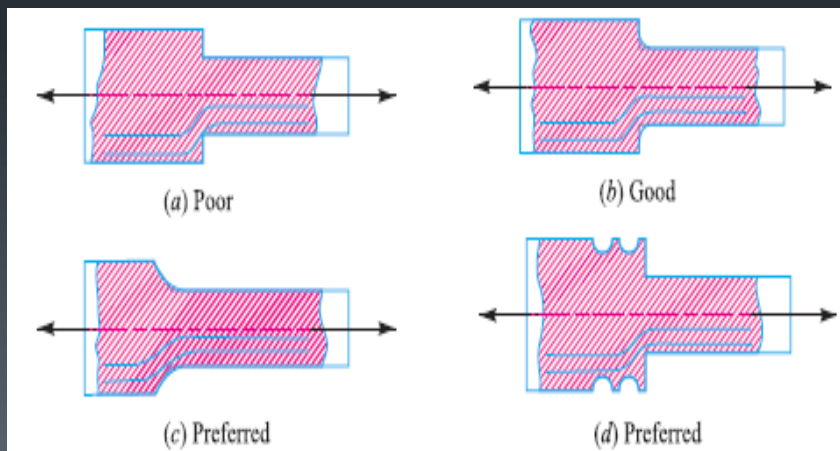
Methods of Reducing Stress Concentration

- Avoiding sharp corners
- Providing fillets
- Use of multiple holes instead of single hole.
- Undercutting the shoulder part

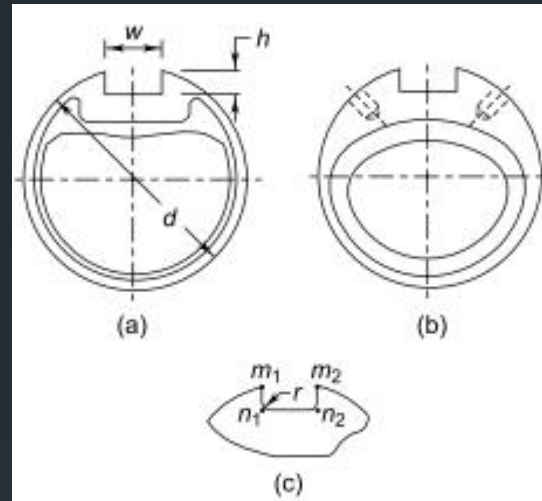
Additional Notches and Holes in Tension Member



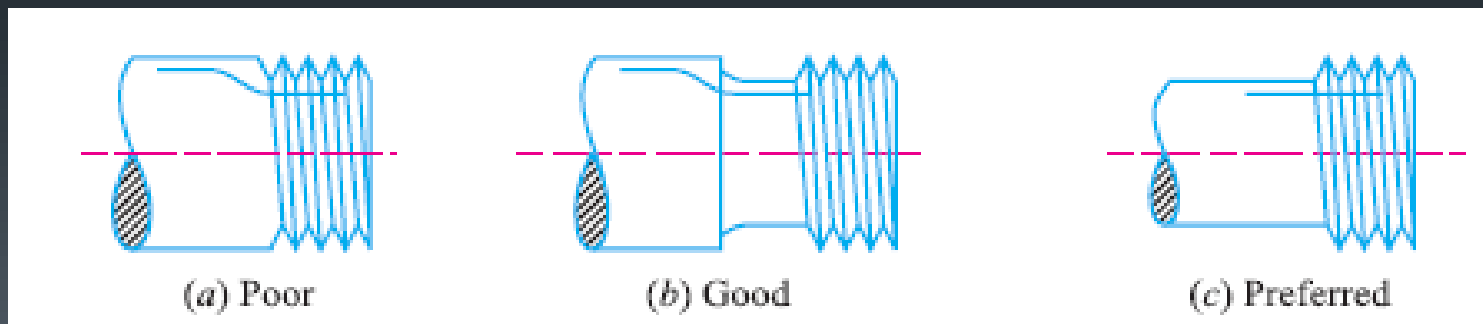
Fillet Radius, Undercutting and Notch for Member in Bending:



▪ *Drilling Additional Holes for Shaft:*



▪ *Reduction of Stress Concentration in Threaded Members:*



Stress Concentration Factors

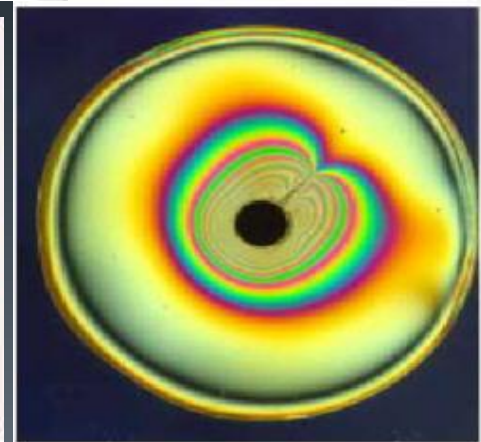
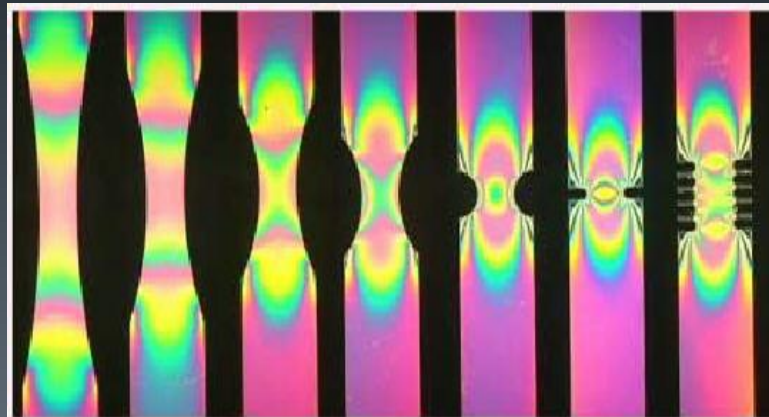
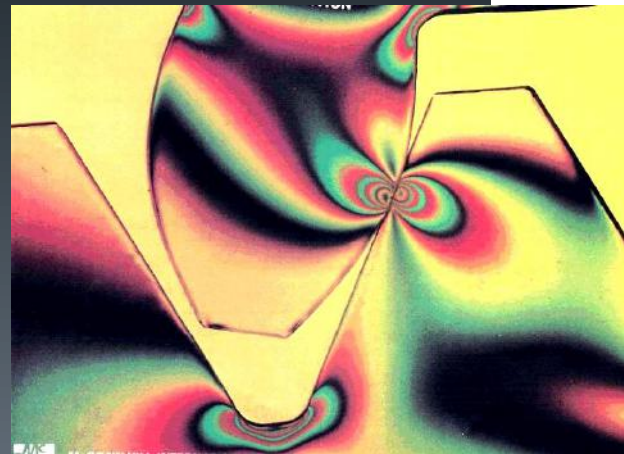
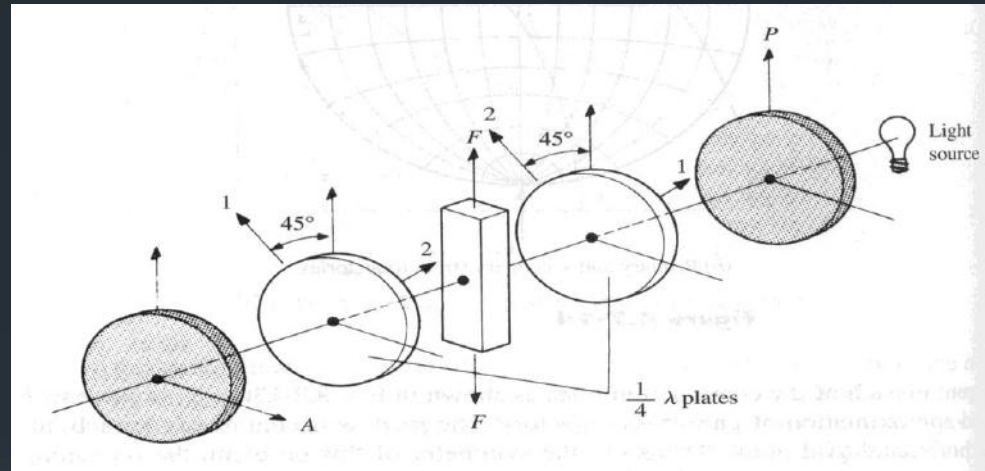


Stress concentration factors are determined by

- Mathematical Model
- Photo-elasticity method
- Brittle Coatings
- Electrical strain gauges
- Finite element method

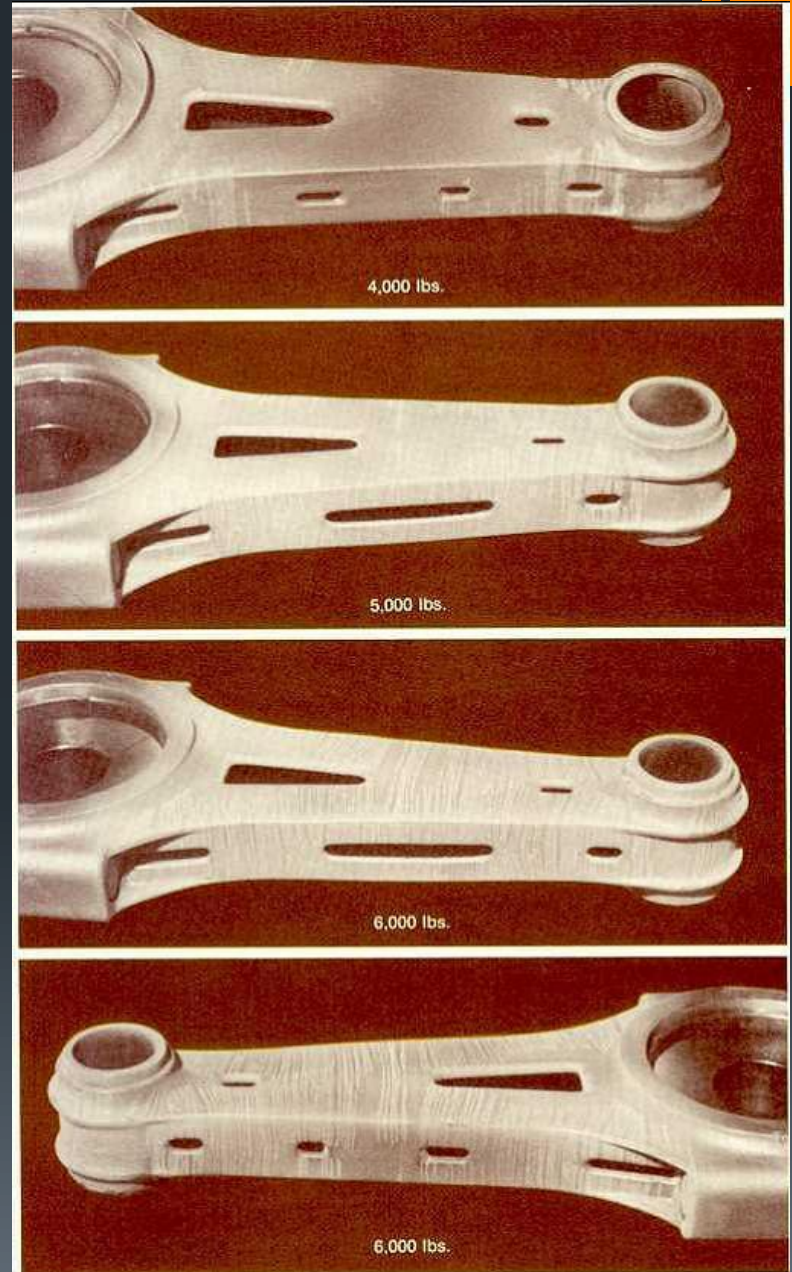
Photoelasticity

A plane polarized light is passed thru a photoelastic material (all transparent plastics) resulting in a colorful fringe pattern indicating the intensity of the stress.



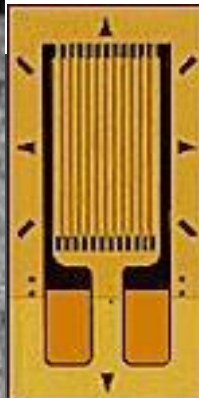
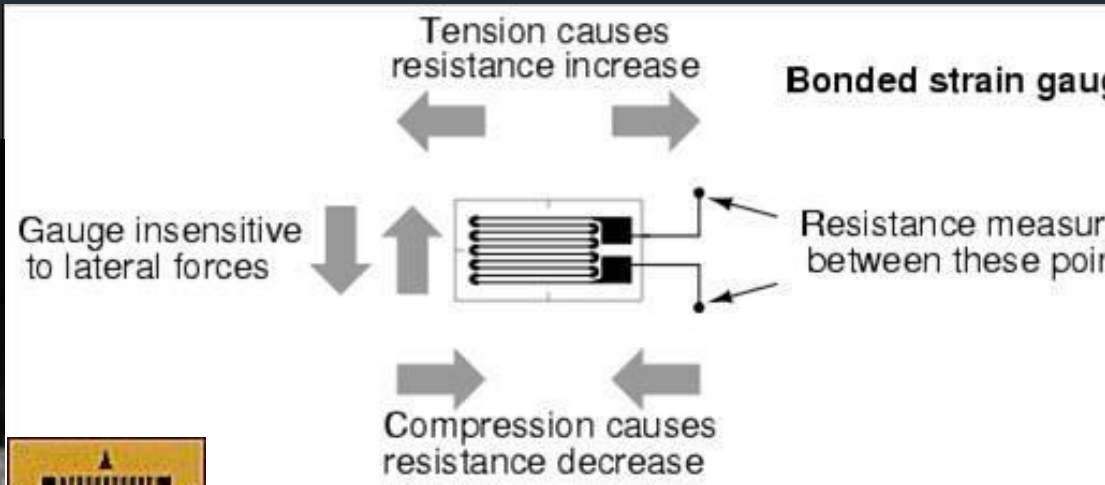
Coatings

Brittle Coating – a specially prepared lacquers are usually applied by spraying on the actual part. After air drying, the part is subjected to stress. A pattern of small cracks appear on the surface. Data could be used to locate strain gages for precise measurement of the stress. The method is sensitive to temperature and humidity.



Electrical Strain Guages

The method is the most popular and widely accepted for strain measurements and stress analysis. The strain gauge consists of a grid of strain-sensitive metal foil bonded to a plastic backing material. When the gauge is subjected to a mechanical deformation, its electrical resistance changes proportionally. The change in voltage is converted to strain and the stress is calculated from the strain.



Finite Element method

- Pre-processor
- Solution
- Post-processor



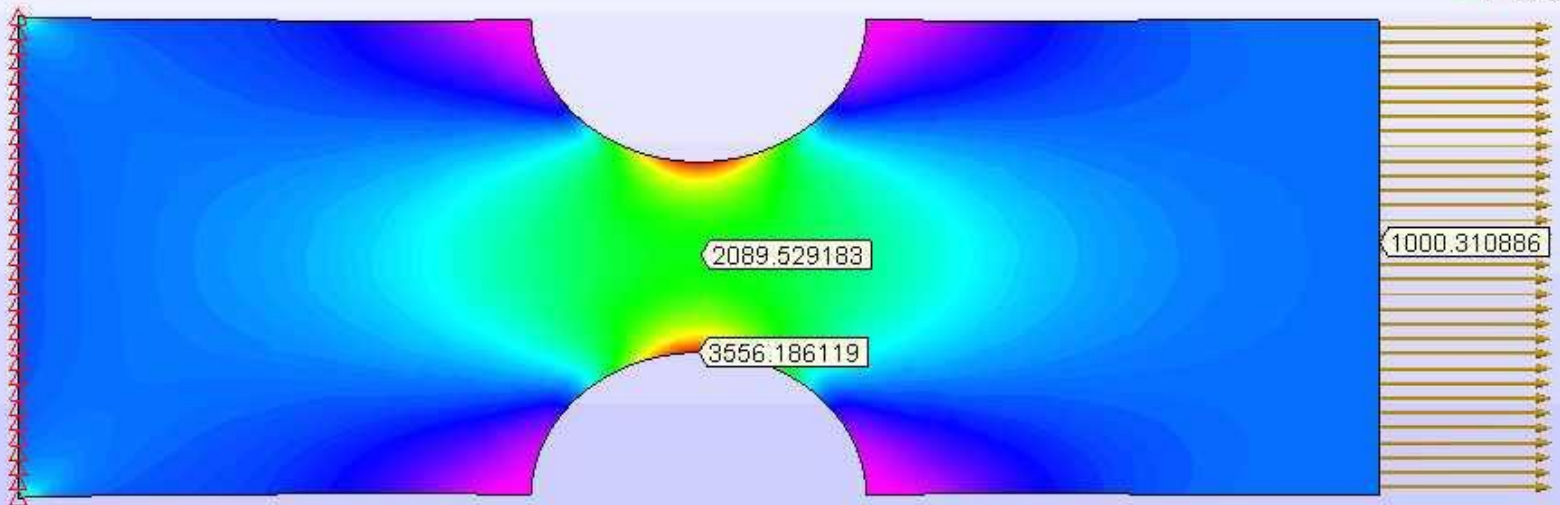
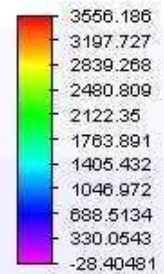
• CAD Assembly with Loads and Supports

• Meshed Assembly (Hex and Tet Elements)

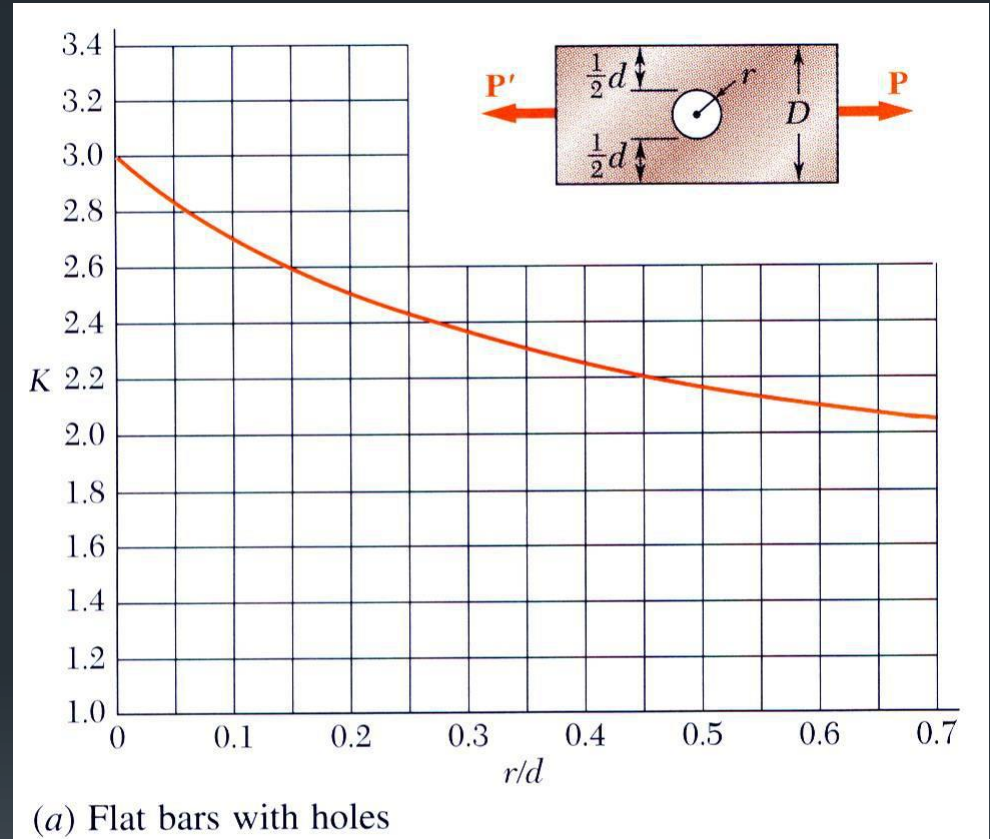
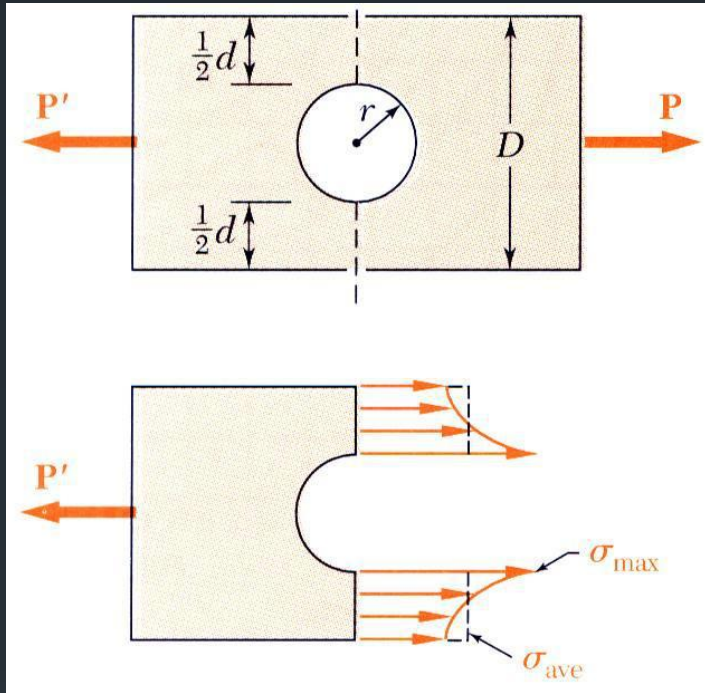
• Stress Plot of Solved Problem

$$K = \frac{\sigma_{\max}}{\sigma_{\text{ave}}} = \frac{3556}{2090} = 1.701$$

Stress
Tensor Y-Y
lbf/(in²)



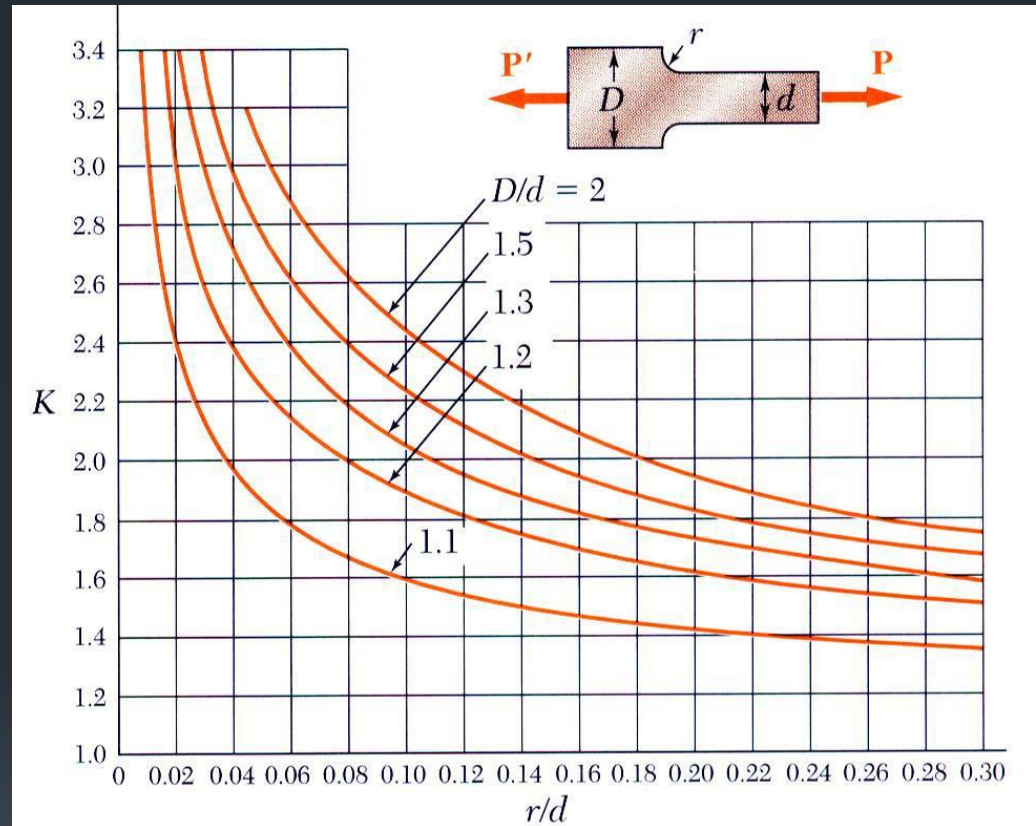
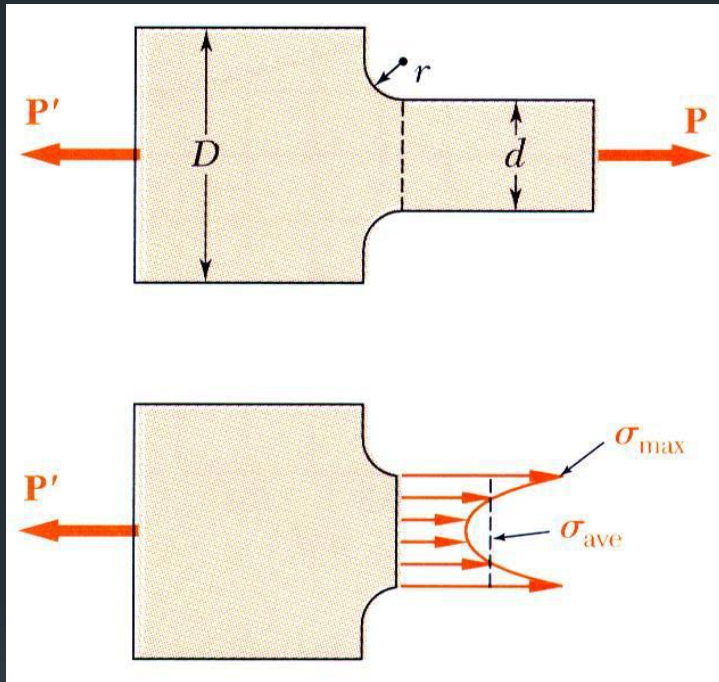
Stress Concentration: Hole



Discontinuities of cross section may result in high localized or *concentrated* stresses.

$$K = \frac{\sigma_{\max}}{\sigma_{\text{ave}}}$$

Stress Concentration: Fillet



(b) Flat bars with fillets

Fatigue:

When a material is subjected to repeated stresses, it fails at stresses below the yield point stresses. Such type of failure of a material is known as ***fatigue**.

The failure is caused by means of a progressive crack formation which are usually fine and of microscopic size. This property is considered in designing shafts, connecting rods, springs, gears, etc

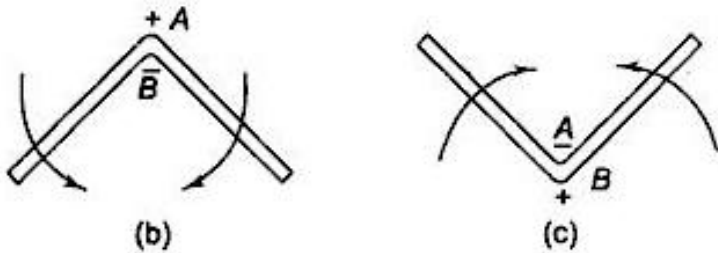
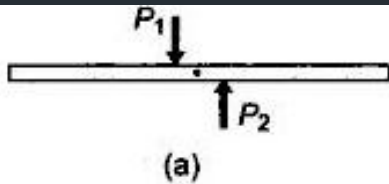
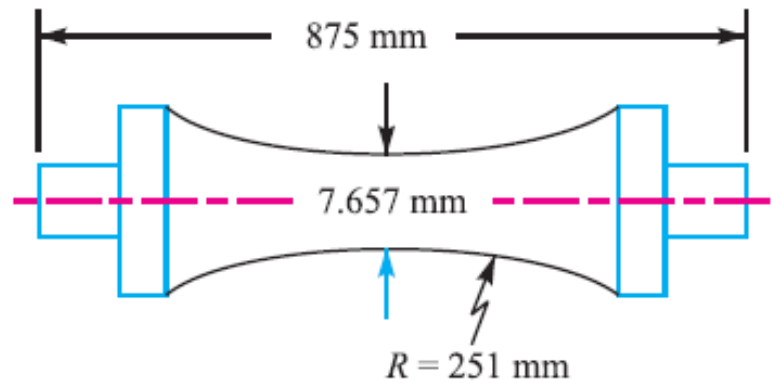


Fig. 5.16 Shear and Fatigue Failure of Wire:
(a) Shearing of Wire (b) Bending of Wire
(c) Unbending of Wire

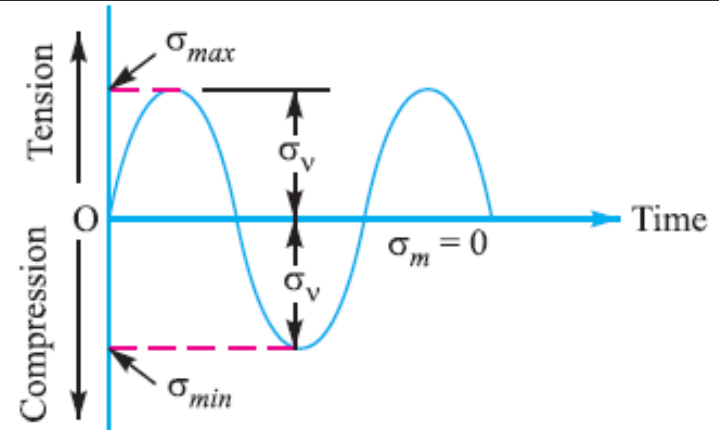
- (i) Regions of discontinuity, such as oil holes, keyways, screw threads, etc.
- (ii) Regions of irregularities in machining operations, such as scratches on the surface, stamp mark, inspection marks, etc.
- (iii) Internal cracks due to defects in materials like blow holes

Endurance limit by rotating beam method

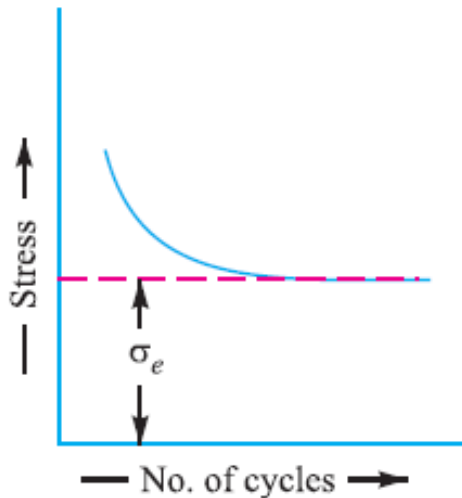




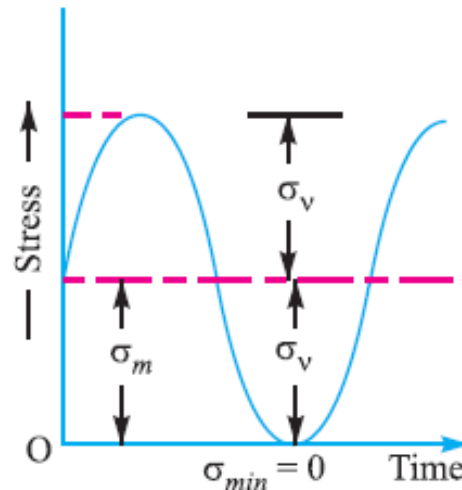
(a) Standard specimen.



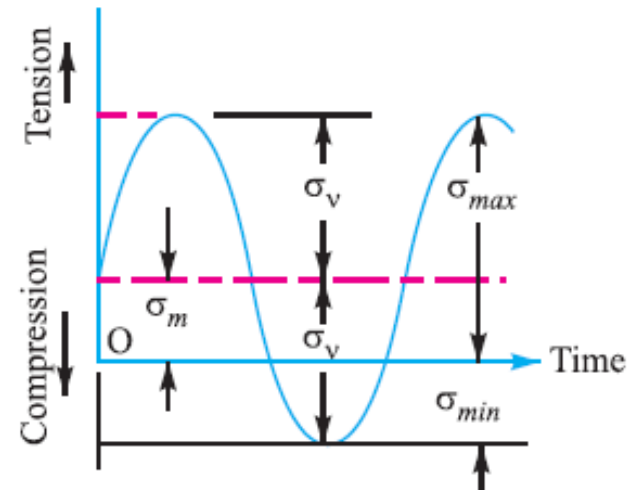
(b) Completely reversed stress.



(c) Endurance or fatigue limit.



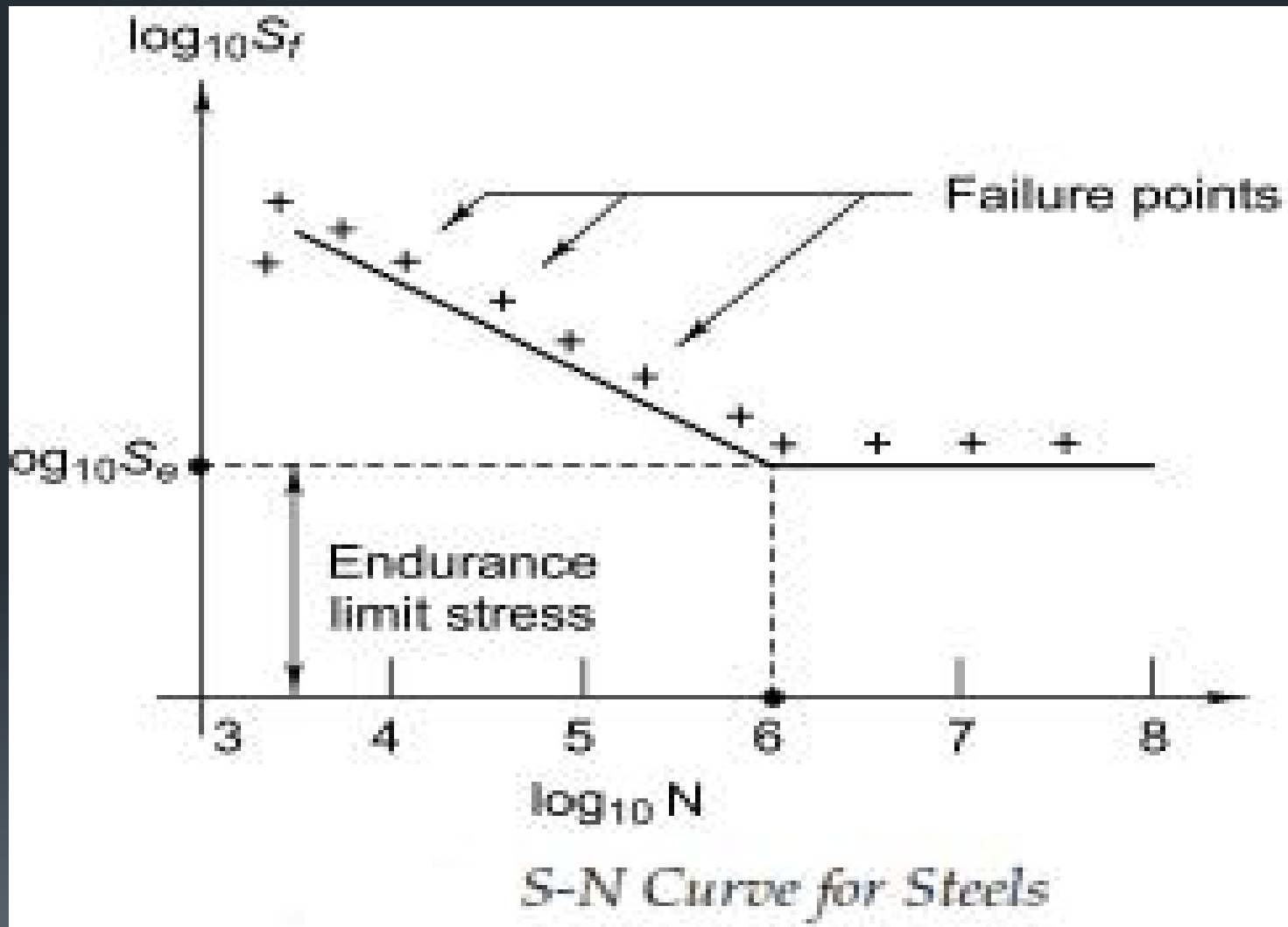
(d) Repeated stress.



(e) Fluctuating stress.

Fig. 6.2. Time-stress diagrams.

S-N Curve or Wohler diagram



ENDURANCE LIMIT

- It may be defined as the safe maximum stress which can be applied to the machine part working under actual condition.
- It is defined as maximum value of completely reversed bending stress which a polished specimen can withstand without failure for infinite number of cycles.

FACTORS AFFECTING ENDURANCE STRENGTH

- Load factor (K_L)
- Surface finish factor(K_{SF})
- Size factor (K_{SZ})
- Reliability factor(K_R)
- Miscellaneous factors(K) ns.

NOTCH SENSITIVITY (q)

This is defined as the degree to which the actual stress concentration effect compares with theoretical stress concentration effect.

Effect of Load factor (K_L)

- Shigley and Mischke proposed following exponential formulae to calculate load factor

$$\text{For } 2.79 \text{ mm} \leq d < 51 \text{ mm} \\ K_b = 1.24 d^{-0.107}$$

$$\text{For } 51 \text{ mm} < d \leq 254 \text{ mm} \\ K_b = 0.859 - 0.000873 d$$

\therefore Endurance limit for reversed bending load, $\sigma_{eb} = \sigma_e \cdot K_b = \sigma_e$
Endurance limit for reversed axial load, $\sigma_{ea} = \sigma_e \cdot K_a$
and endurance limit for reversed torsional or shear load, $\tau_e = \sigma_e \cdot K_s$

Load factor : $K_a=0.8$

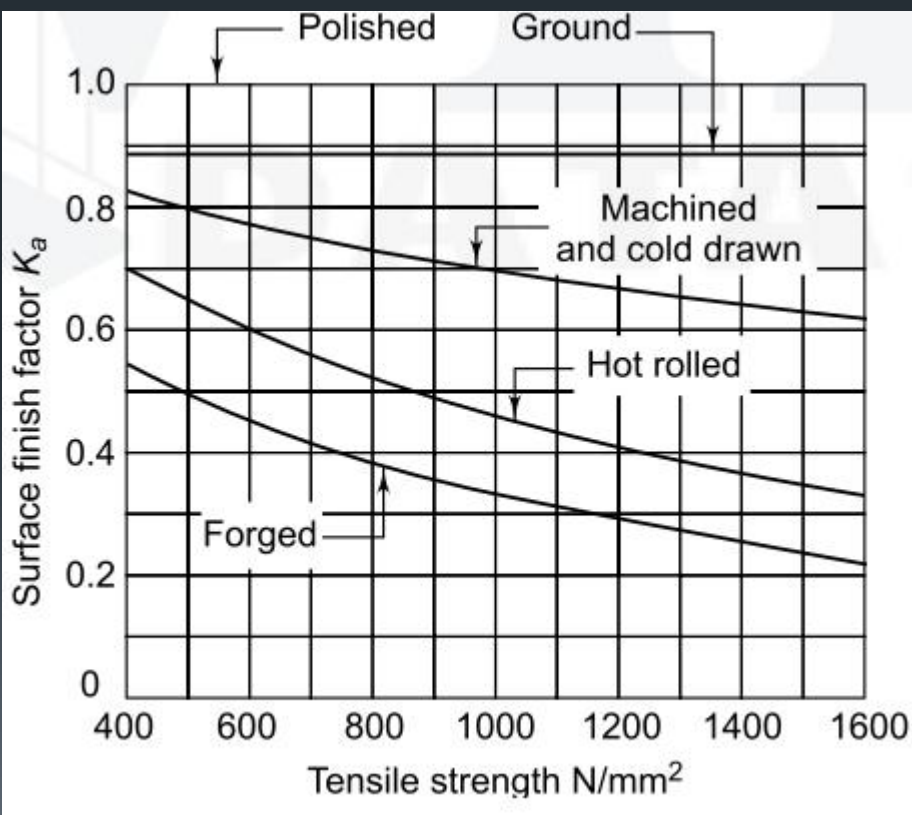
$k_s=0.55$ for ductile 0.8 for brittle

Surface Finish Factor (K_{SF})

- Shigley and Mischke have suggested an exponential relation for surface

$$K_a = a(S_{ut})^b \quad [\text{if } K_a > 1, \text{ set } K_a = 1]$$

Surface finish	a	b
Ground	1.58	-0.085
Machined or cold-drawn	4.51	-0.265
Hot-rolled	57.7	-0.718
As forged	272	-0.995



Let K_{sur} = Surface finish factor.
 \therefore Endurance limit,
 $\sigma_{e1} = \sigma_{eb} \cdot K_{sur} = \sigma_e \cdot K_b \cdot K_{sur} = \sigma_e \cdot K_{sur}$... ($\because K_b = 1$)
 ... (For reversed bending load)
 $= \sigma_{ea} \cdot K_{sur} = \sigma_e \cdot K_a \cdot K_{sur}$... (For reversed axial load)
 $= \tau_e \cdot K_{sur} = \sigma_e \cdot K_s \cdot K_{sur}$... (For reversed torsional or shear load)

te : The surface finish factor for non-ferrous metals may be taken as unity.

Size Factor (K_{sz})

Let K_{sz} = Size factor.

∴ Endurance limit,

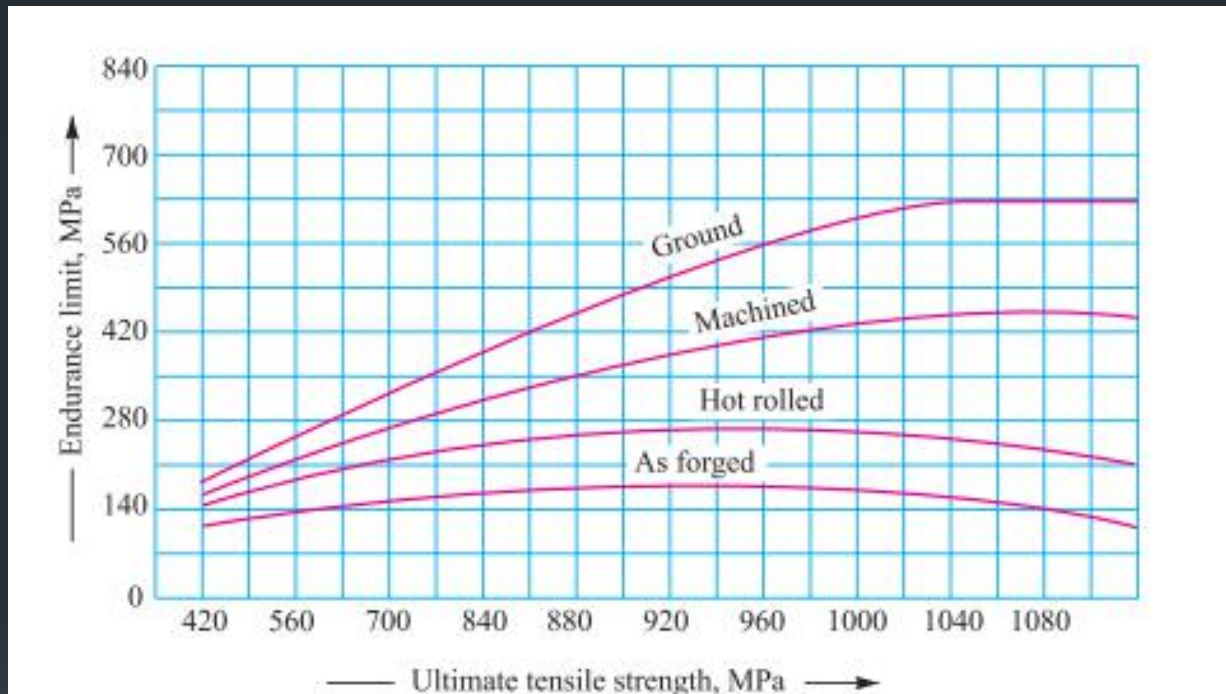
$$\begin{aligned}\sigma_{e2} &= \sigma_{e1} \times K_{sz} && \dots(\text{Considering surface finish factor also}) \\ &= \sigma_{eb} \cdot K_{sur} \cdot K_{sz} = \sigma_e \cdot K_b \cdot K_{sur} \cdot K_{sz} = \sigma_e \cdot K_{sur} \cdot K_{sz} && (\because K_b = 1) \\ &= \sigma_{ea} \cdot K_{sur} \cdot K_{sz} = \sigma_e \cdot K_a \cdot K_{sur} \cdot K_{sz} && \dots(\text{For reversed axial load}) \\ &= \tau_e \cdot K_{sur} \cdot K_{sz} = \sigma_e \cdot K_s \cdot K_{sur} \cdot K_{sz} && \dots (\text{For reversed torsional or shear load})\end{aligned}$$

Notes: 1. The value of size factor is taken as unity for the standard specimen having nominal diameter of 7.657 mm.

2. When the nominal diameter of the specimen is more than 7.657 mm but less than 50 mm, the value of size factor may be taken as 0.85.

3. When the nominal diameter of the specimen is more than 50 mm, then the value of size factor may be taken as 0.75.

Relation between endurance strength and ultimate stress



For steel, $\sigma_e = 0.5 \sigma_u$;

For cast steel, $\sigma_e = 0.4 \sigma_u$;

For cast iron, $\sigma_e = 0.35 \sigma_u$;

For non-ferrous metals and alloys, $\sigma_e = 0.3 \sigma_u$

Reliability factor(K_R)

- The reliability factor is depends upon the reliability requirement of the mechanical component.
- The reliability correction factor accounts for the scatter and uncertainty of material properties (endurance limit).

Reliability	50	90	95	99	99.9	99.99	99.999
Factor	1	0.897	0.868	0.814	0.753	0.702	0.659

Miscellaneous factors(K) ns.

1. For the reversed bending load, endurance limit,

$$\sigma'_e = \sigma_{eb} \cdot K_{sur} \cdot K_{sz} \cdot K_r \cdot K_f \cdot K_i$$

2. For the reversed axial load, endurance limit,

$$\sigma'_e = \sigma_{ea} \cdot K_{sur} \cdot K_{sz} \cdot K_r \cdot K_f \cdot K_i$$

3. For the reversed torsional or shear load, endurance limit,

$$\sigma'_e = \tau_e \cdot K_{sur} \cdot K_{sz} \cdot K_r \cdot K_f \cdot K_i$$

In solving problems, if the value of any of the above factors is not known, it may be taken as unity.

Modifying factor for stress concentration

$$k_d = 1/k_f$$

Fatigue stress concentration factor

- When a machine member is subjected to cyclic or fatigue loading, the value of fatigue stress concentration factor shall be applied instead of theoretical stress concentration factor.

$$K_f = \frac{\text{Endurance limit without stress concentration}}{\text{Endurance limit with stress concentration}}$$

NOTCH SENSITIVITY (q)

This is defined as the degree to which the actual stress concentration effect compares with theoretical stress concentration effect.

The notch sensitivity factor q is defined as

$$q = \frac{\text{Increase of actual stress over nominal stress}}{\text{Increase of theoretical stress over nominal stress}}$$

$$q = \frac{K_f - 1}{K_t - 1}$$

$$K_f = 1 + q (K_t - 1)$$

$$K_{fs} = 1 + q (K_{ts} - 1)$$

REVERSED STRESSES— DESIGN FOR FINITE AND INFINITE LIFE

There are two types of problems in fatigue design—

- (i) Components subjected to completely reversed stresses
 - Design for infinite life
 - Design for finite life.
- (ii) Components subjected to fluctuating stresses.

Design for infinite life

- When the component is to be designed for infinite life, the endurance limit becomes the criterion of failure.
- The amplitude stress induced in such components should be lower than the endurance limit in order to withstand the infinite number of cycles.
- Such components are designed with the help of the following equations:

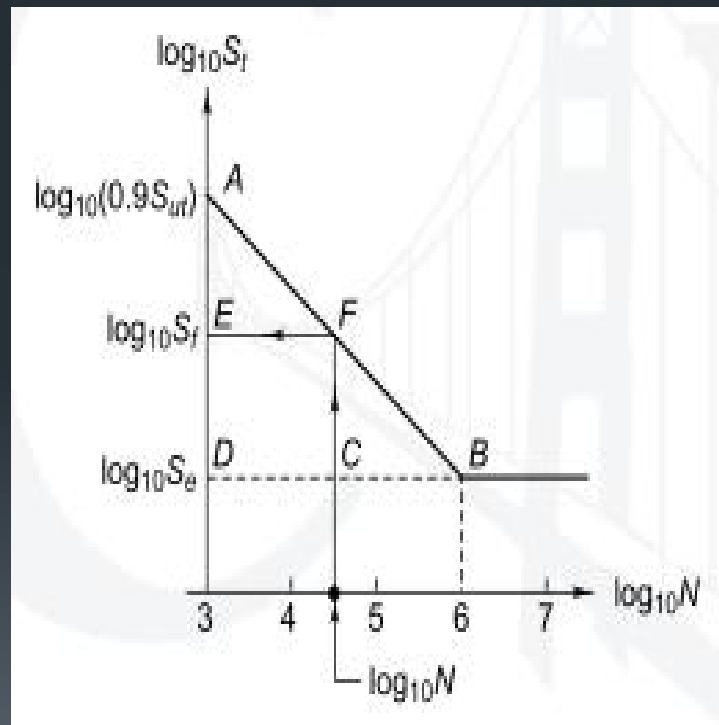
$$\sigma_a = \frac{S_e}{(f_s)}$$

$$\tau_a = \frac{S_{se}}{(f_s)}$$

Design for finite life

- When the design is based on finite life S-N curve is used.

It consists of a straight line AB drawn from $(0.9\sigma_{ut})$ at 10^3 cycles to σ_e at 10^6 cycles on log-log paper.



Design procedure for finite life problems

- Locate the point A with co-ordinates $[3, \log_{10}(0.9\sigma_{ut})]$, since
 $\log_{10}(10^3)=3$
- Locate the point B with co-ordinates $[6, \log_{10}(\sigma_e)]$, since
 $\log_{10}(10^6)=6$
- Join \overline{AB} , Which is used as a criterion of failure for finite life problems
- Depending upon the life N of the component. Draw a vertical line passing through $\log_{10}(N)$ on the abscissa. This line intersects \overline{AB} , at point F.
- Draw a line \overline{FE} parallel to the abscissa. The ordinate at point E, i.e. $\log_{10}(\sigma_f)$ gives the fatigue strength corresponding to N cycles.

Q. A rotating bar made of steel 45C8 ($S_{ut} = 630 \text{ N/mm}^2$) is subjected to a completely reversed bending stress. The corrected endurance limit of the bar is 315 N/mm^2 . Calculate the fatigue strength of the bar for a life of 90,000 cycles.

Given $S_{ut} = 630 \text{ N/mm}^2$ $S_e = 315 \text{ N/mm}^2$
 $N = 90000 \text{ cycles}$

Step I Construction of $S-N$ diagram

$$0.9S_{ut} = 0.9 (630) = 567 \text{ N/mm}^2$$

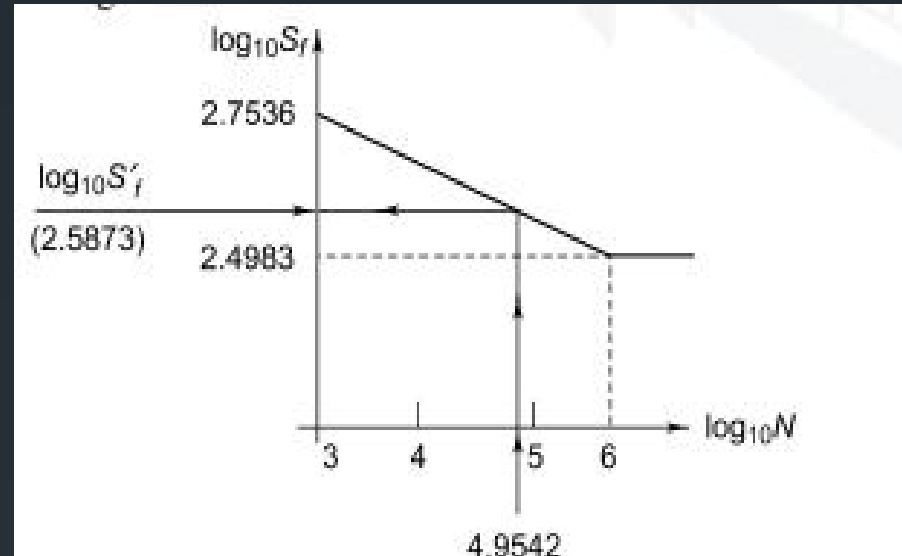
$$\log_{10} (0.9S_{ut}) = \log_{10} (567) = 2.7536$$

$$\log_{10} (S_e) = \log_{10} (315) = 2.4983$$

$$\log_{10} (90\,000) = 4.9542$$

$$\text{Also, } \log_{10} (10^3) = 3 \quad \text{and} \quad \log_{10} (10)^6 = 6$$

Figure 5.30 shows the $S-N$ curve for the bar.



Step II Fatigue strength for 90000 cycles

Referring to Fig. 5.30,

$$\log_{10} (S'_f) = 2.7536 - \frac{(2.7536 - 2.4983)}{(6 - 3)}$$

$$\times (4.9542 - 3) = 2.5873$$

$$S'_f = 386.63 \text{ N/mm}^2$$





Design of Keys

Presented by

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Unit 4 Design of Keys, Cotter and Knuckle Joints

Keys: Design of sunk, saddle, Kennedy and woodruff keys.

Cotter and Knuckle Joints: Socket and spigot joint, Sleeve and Cotter joints, Gib and Cotter joint, Knuckle joint.

Course Outcome 4:

At the end of the topic, student will be able to **design keys, cotter and knuckle joints for various applications.**

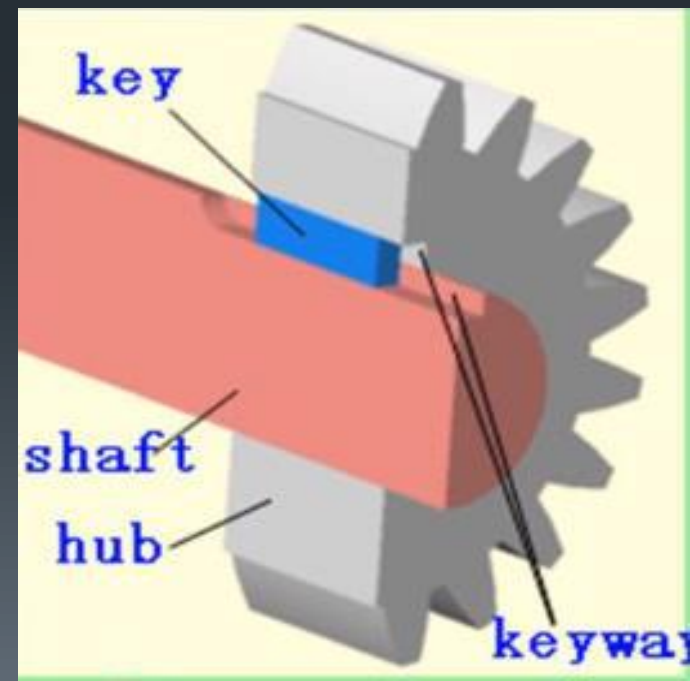
Key

A key is a fastening inserted into the keyway of two mating parts, it is used to transmit from a shaft to a hub or vice versa.

- Keys are used as temporary fastenings and are subjected to considerable crushing and shearing stresses.
- It is inserted parallel to the axis of the shaft in a groove or slot which called “**keyway**”.

Functions of Keys:

- To transmit power from a shaft to hub or vice versa
- To prevent relative rotational motion between the shaft and the joined machine element like gear or pulley.



Types of Keys

Shaft keys come in a wide variety of types and shapes and can be divided into four categories and subcategories.

- Sunk Keys
 - Rectangular & square keys
 - Parallel keys
 - Gib head keys
 - Feather key (sliding clearance with keys)
 - Woodruff key
- Saddle keys
 - Flat & Hollow saddle keys
- Tangent keys
- Round/Circular keys

Sunk Keys

- The sunk keys are provided half in the keyway of the shaft and half in the keyway of the hub or boss of the pulley.

The types of sunk keys are as follows

1. Rectangular sunk key
2. Parallel keys
3. Gib head keys
4. Feather key (sliding clearance with keys)
5. Woodruff key

Rectangular sunk key

A rectangular sunk key is shown in Fig.

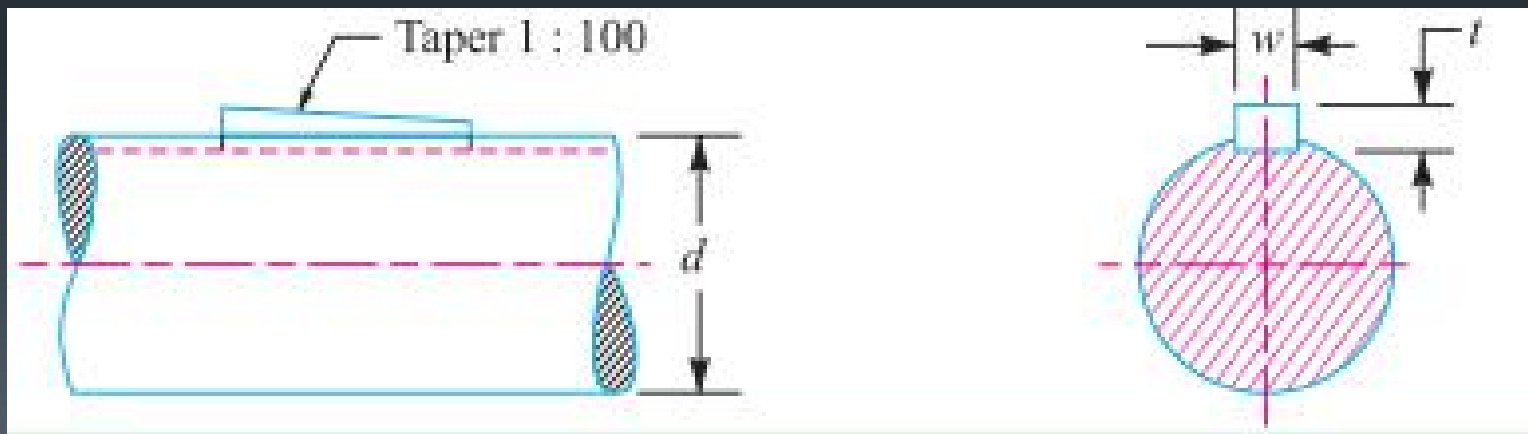
The usual proportions of this key are :

Width of key, $w = d / 4$; and

thickness of key, $t = 2w / 3 = d / 6$

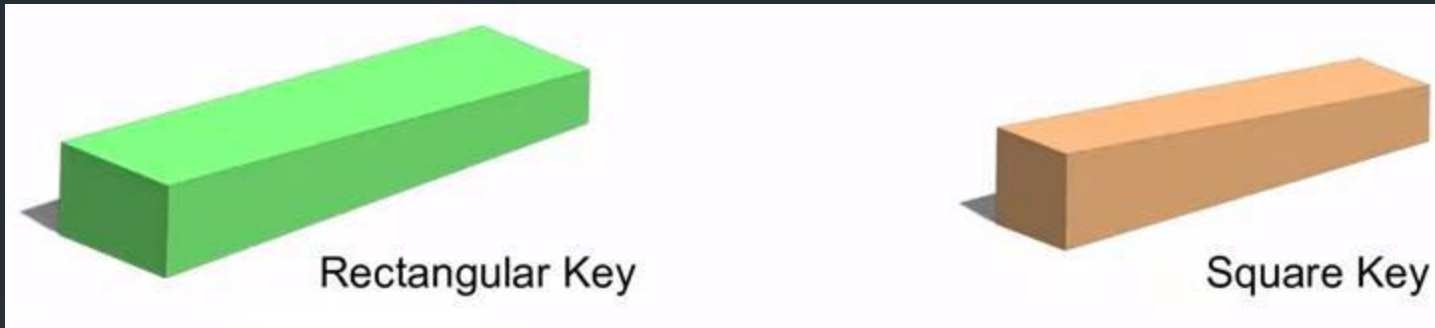
where $d =$ Diameter of the shaft or diameter of the hole in the hub.

The key has taper 1 in 100 on the top side only.



Parallel sunk key

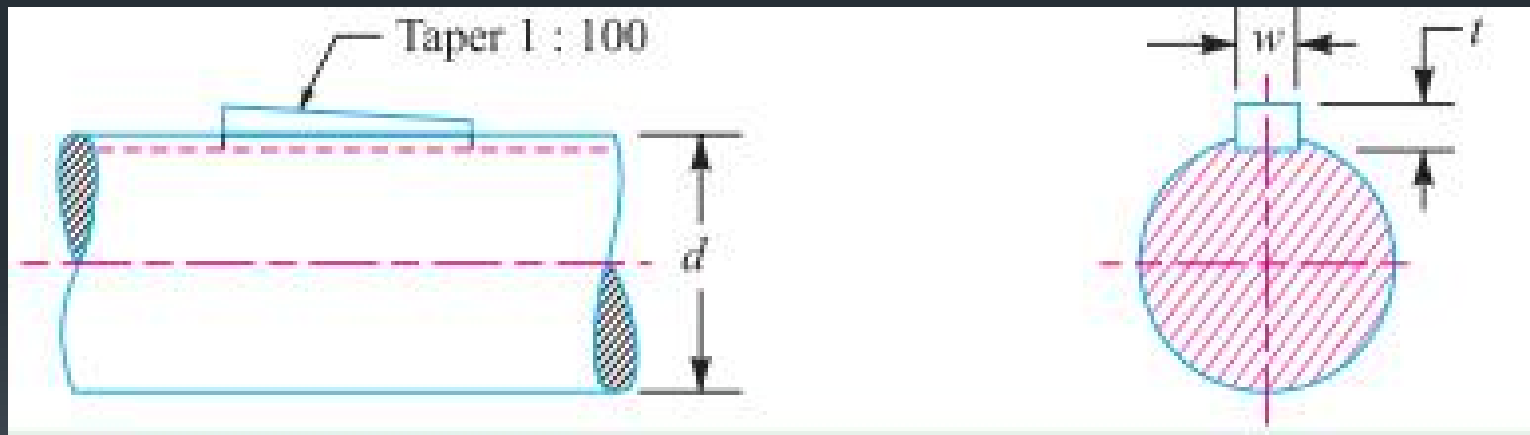
It is a taperless Rectangular or square sunk key is shown in Fig.
Uses: Hub is required to slide along the shaft.



Square Key

- The only difference between a rectangular sunk key and a square sunk key is that its width and thickness are equal.

$$w = t = d / 4$$



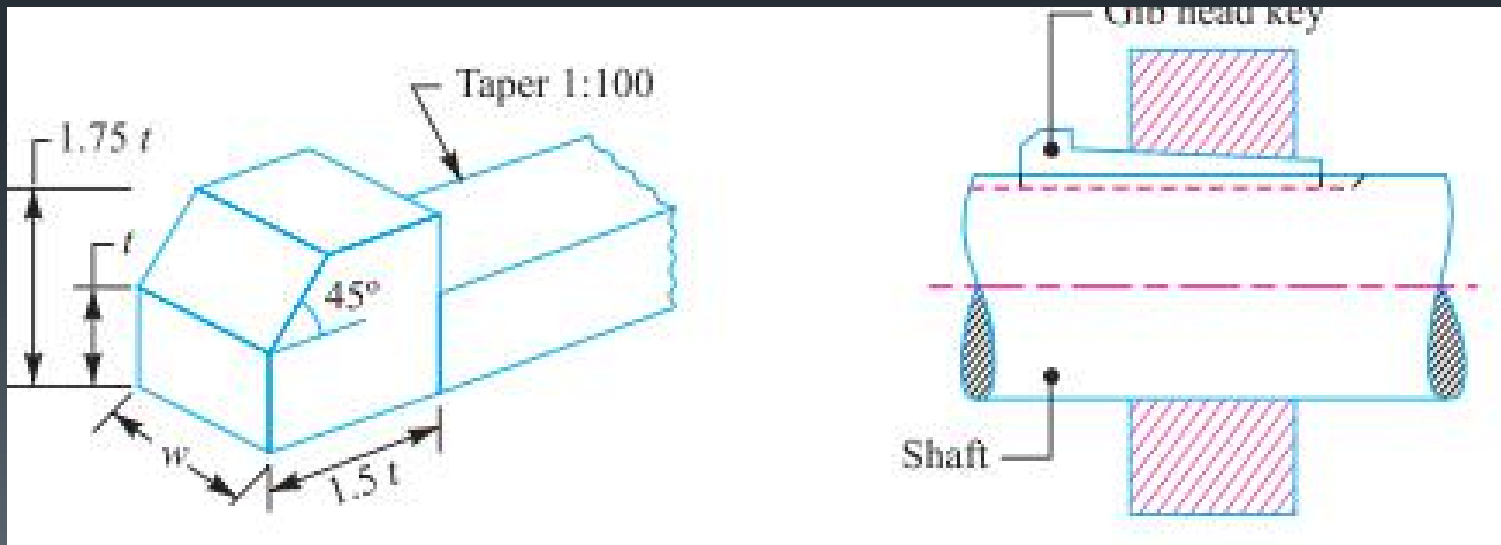
Gib Head Key

It is a rectangular sunk key with a head at one end known as gib head.

- It is usually provided to facilitate the removal of key. A gib head key is shown in Fig.
- The usual proportions of the gib head key are :

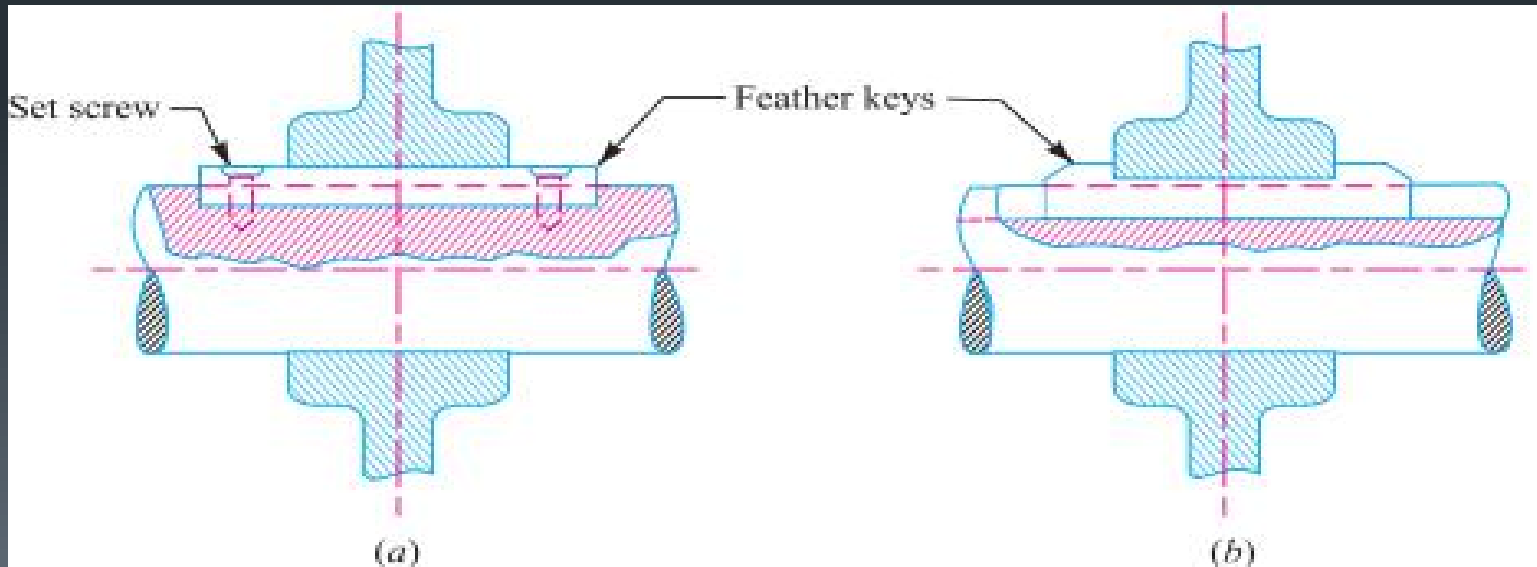
Width, $w = d / 4$;

and thickness at large end, $t = 2w / 3 = d / 6$

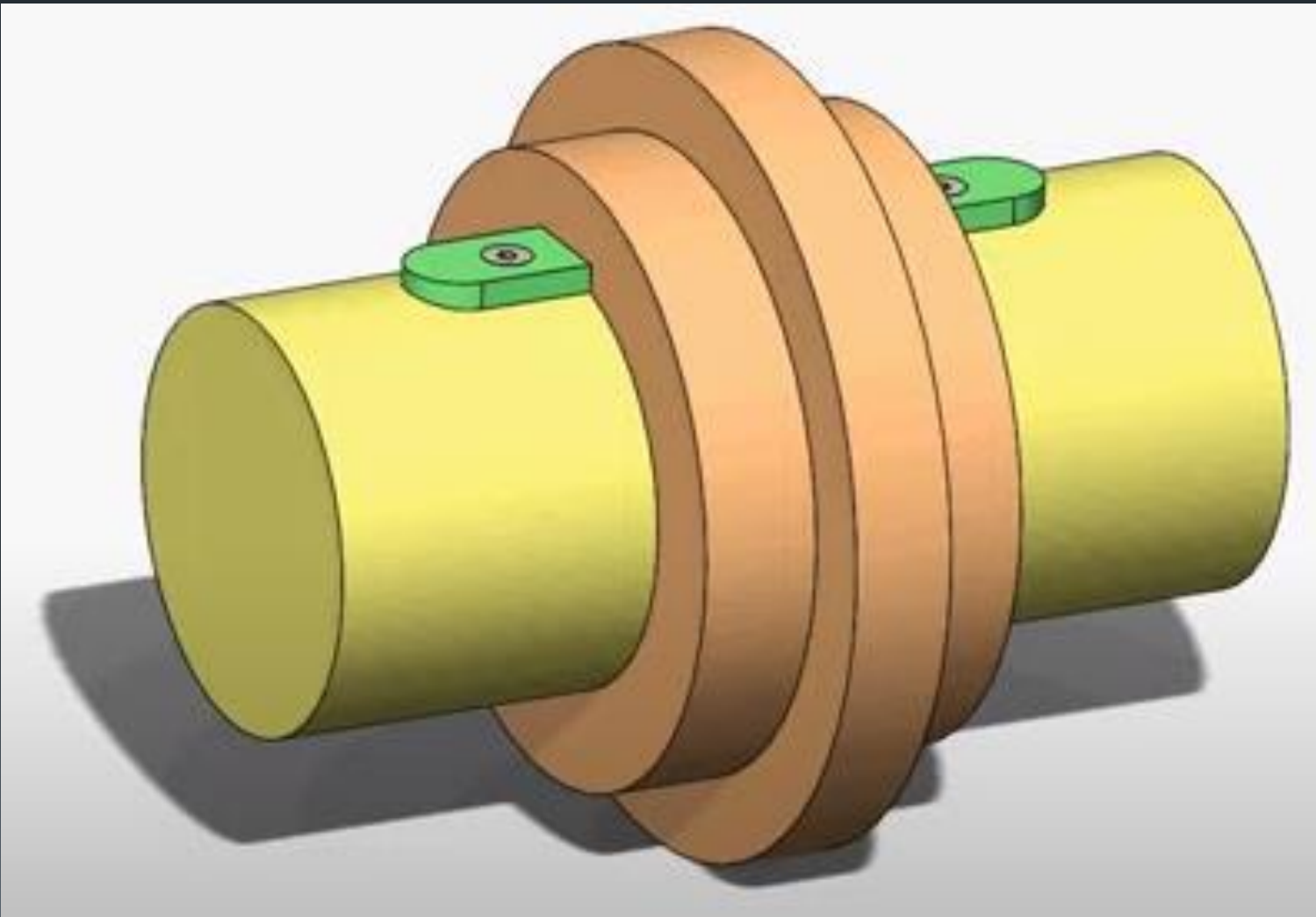


Feather Key

- It is fitted either to the shaft or Hub
- A key attached to one member of a pair and which permits **relative axial movement** is known as *feather key*. *It is a special type of parallel key which transmits a turning moment and also permits axial movement.*
- It is fastened either to the shaft or hub, the key being a sliding fit in the key way of the moving piece.



Cont..



Wood Ruff key

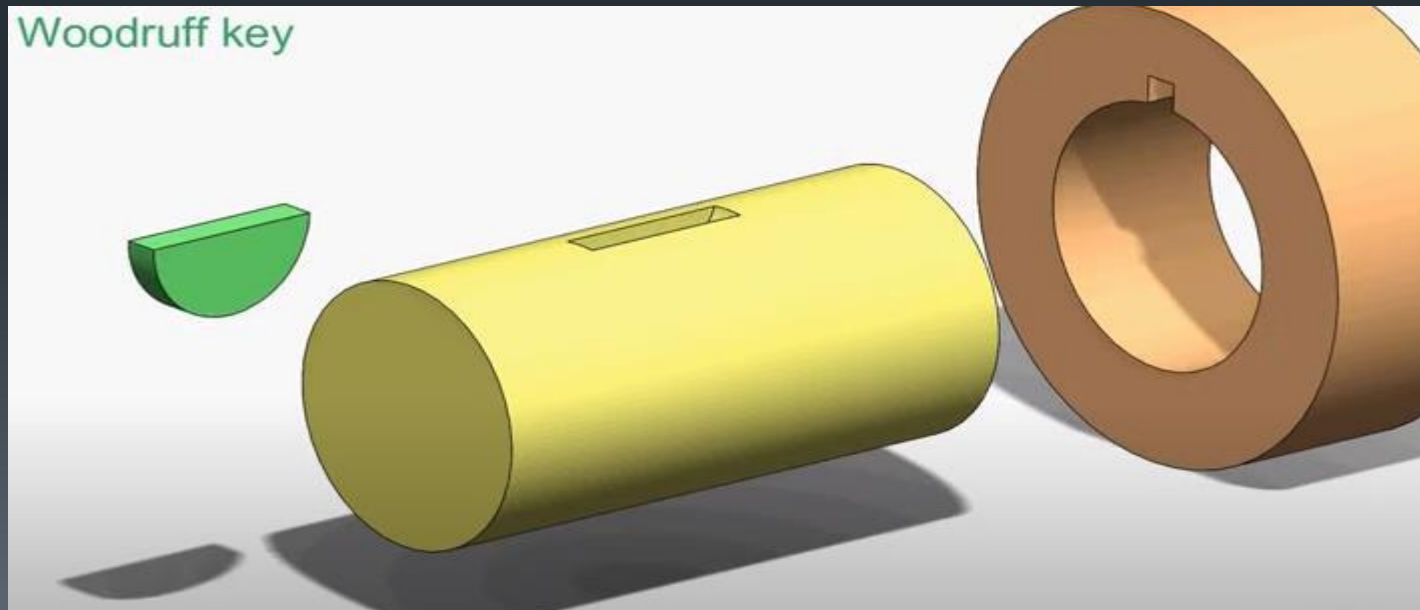
The woodruff key is an easily adjustable key. It is a piece from a cylindrical disc having segmental cross-section in front view as shown in Fig

- **Advantages:** Accommodates itself to any taper in the hub

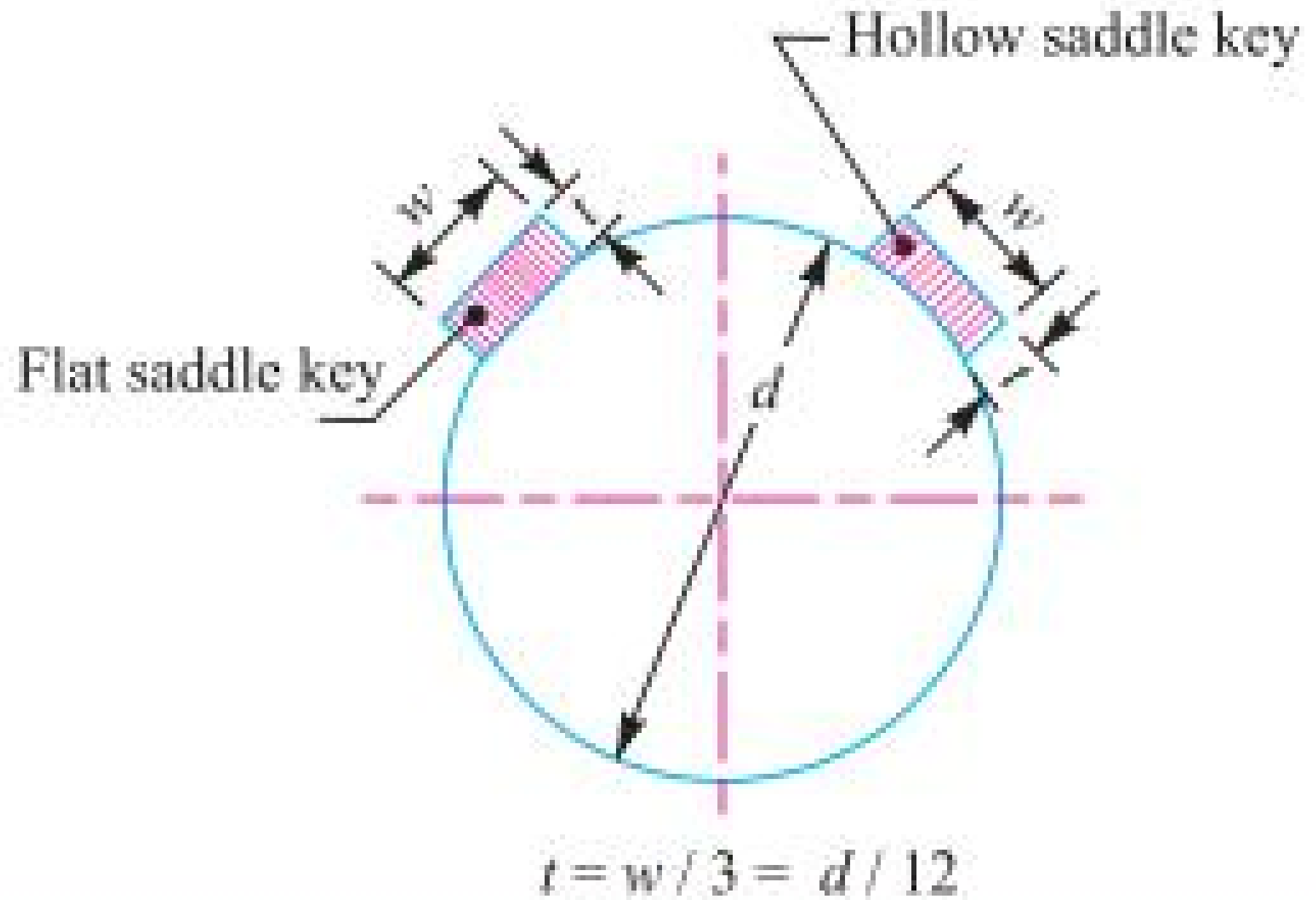
Its extra depth in the shaft, prevents any tendency to turnover in the keyway

- **Disadvantages:** Depth of the keyway weakens the shaft

It cannot be used as a feather key

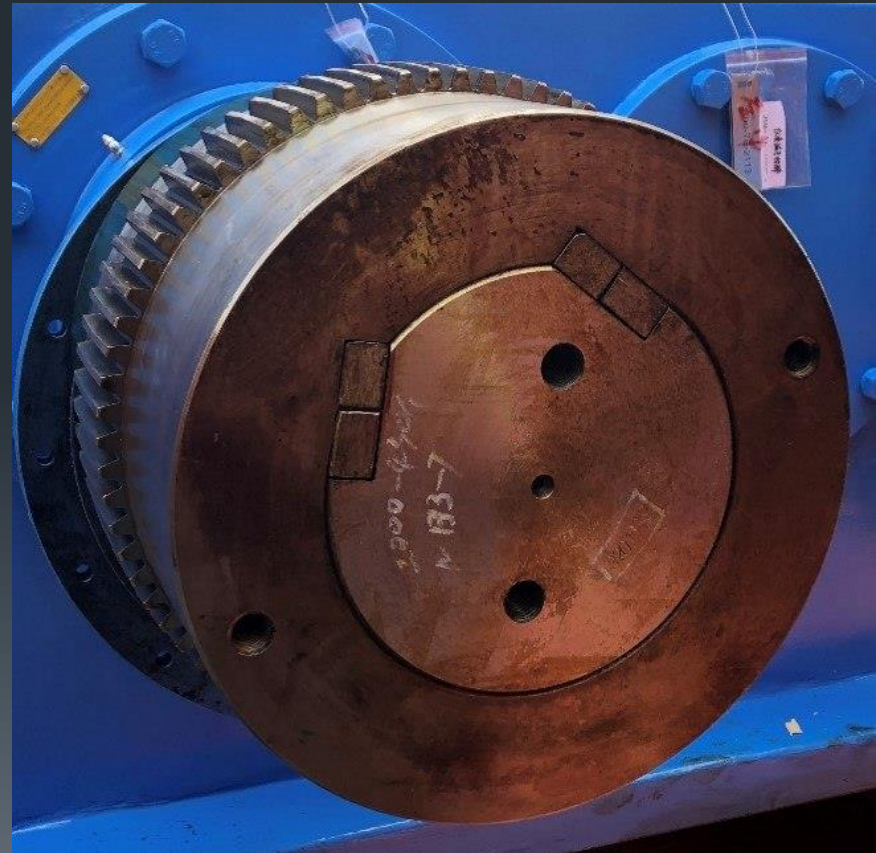
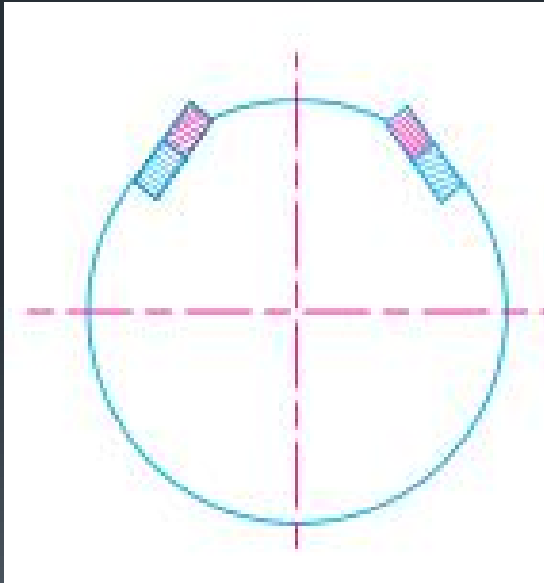


Saddle keys



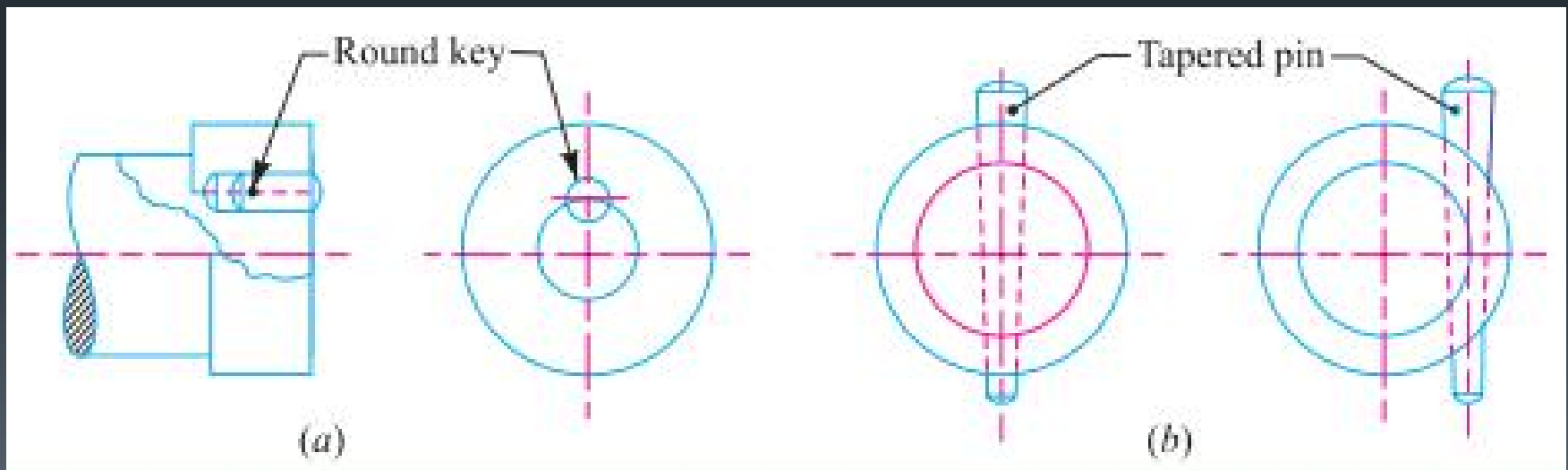
Tangent Key

- The tangent keys are fitted in pair at right angles as shown in Fig. Each key is to withstand torsion in one direction only. These are used in large heavy duty shafts



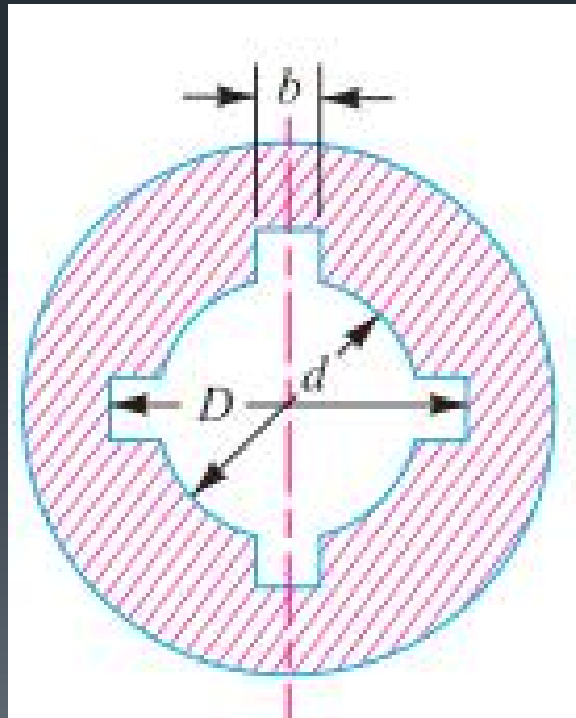
Round Keys

- The round keys, as shown in Fig., *are circular in section and fit into holes drilled partly in the shaft and partly in the hub.* They have the advantage that their keyways may be drilled and reamed after the mating parts have been assembled. Round keys are usually considered to be most appropriate for **low power drives**.



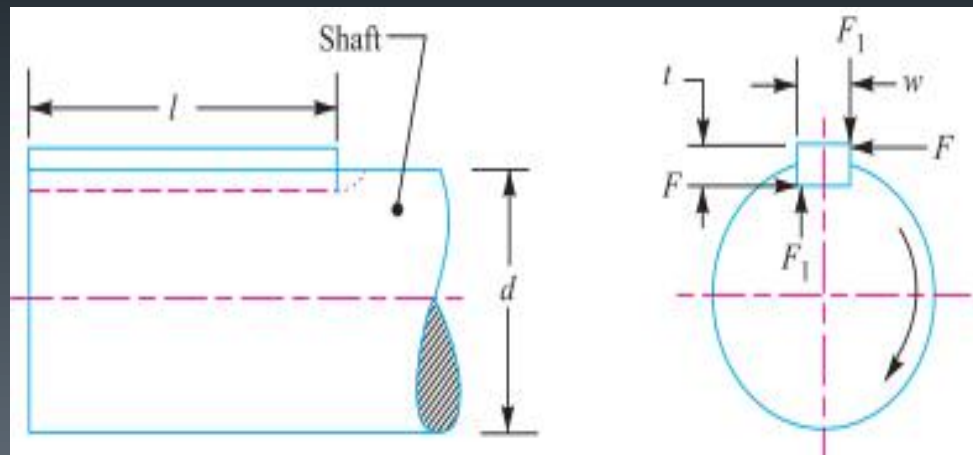
Splines

- **Shaft + Integrated key = Splines**
- The splined shafts are used when the force to be transmitted is large in proportion to the size of the shaft as in automobile transmission and sliding gear transmissions. By using splined shafts, we obtain axial movement as well as positive drive is obtained.



Forces acting on a Sunk Key

- When a key is used in transmitting torque from a shaft to a rotor or hub, the following two types of forces act on the key :
 - Forces (F_1)** due to **fit of the key in its keyway**, as in a tight fitting straight key or in a tapered key driven in place. These forces produce compressive stresses in the key which are difficult to determine in magnitude.
 - Forces (F_2)** due to the **torque transmitted** by the shaft. These forces produce shearing and compressive (or crushing) stresses in the key.



- Considering shearing failure

$$F = \text{Area resisting shearing} \times \text{Shear stress} = l \times w \times \tau$$

$$\text{Torque transmitted by the shaft } T = F \times \frac{d}{2} = l \times w \times \tau \times \frac{d}{2}$$

- Considering crushing failure

$$F = \text{Area resisting crushing} \times \text{Shear stress} = l \times \frac{t}{2} \times \sigma_c$$

$$\text{Torque transmitted by the shaft } T = F \times \frac{d}{2} = l \times \frac{t}{2} \times \sigma_c \times \frac{d}{2}$$

- The key is equally strong in shearing and crushing, if

$$l \times w \times \tau \times \frac{d}{2} = l \times \frac{t}{2} \times \sigma_c \times \frac{d}{2}$$

$$\frac{w}{t} = \frac{\sigma_c}{2\tau}$$

- The permissible crushing stress for the usual key material is at least twice the permissible shearing stress. Therefore from above equation, we have $w = t$. In other words, a square key is equally strong in shearing and crushing.

Design the rectangular key for a shaft of 50 mm diameter. The shearing and crushing stresses for the key material are 42 MPa and 70 MPa.

- Given $d=50$ mm, $\tau=42$ Mpa & $\sigma_c = 70$ Mpa

$$W=d/4=12.5\text{mm}\sim 14\text{mm}$$

$$T=d/6=8.33\text{ mm}\sim 10\text{mm}$$

$$\text{Torque transmitted by shaft } T=\frac{\pi}{16} d^3 \tau = 1.03 \times 10^6 \text{ N-mm}$$

- Considering Shearing failure

$$T=l \times w \times \frac{d}{2} \times \tau$$

$$l=$$

- Considering Crushing failure

$$T=l \times \frac{t}{2} \times \frac{d}{2} \times \sigma_c$$

$$l=$$

A 15 kW, 960 r.p.m. motor has a mild steel shaft of 40 mm diameter and the extension being 75 mm. The permissible shear and crushing stresses for the mild steel key are 56 MPa and 112 MPa. Design the keyway in the motor shaft extension. Check the shear strength of the key against the normal strength of the shaft.

- Given: $P=15000\text{w}$, $N=960\text{ rpm}$, $d=40\text{mm}$, $l = 75\text{ mm}$. $\tau = 56\text{ Mpa}$ & $\sigma_c = 112\text{ Mpa}$

As $\sigma_c = 2\tau$ it is a square key, for which $w=t=d/4=10\text{ mm}$

According to Moore

$$T = \frac{P \times 60}{2 \pi N} = \frac{15 \times 10^3 \times 60}{2 \pi \times 960} = 149\text{ N-m} = 149 \times 10^3\text{ N-mm}$$

$$e = 1 - 0.2 \left(\frac{w}{d} \right) - 1.1 \left(\frac{h}{d} \right) = 1 - 0.2 \left(\frac{w}{d} \right) - 1.1 \left(\frac{t}{2d} \right) = 1 - 0.2 \left(\frac{10}{20} \right) - \left(\frac{10}{2 \times 40} \right) = 0.8125$$

∴ Strength of the shaft with keyway,

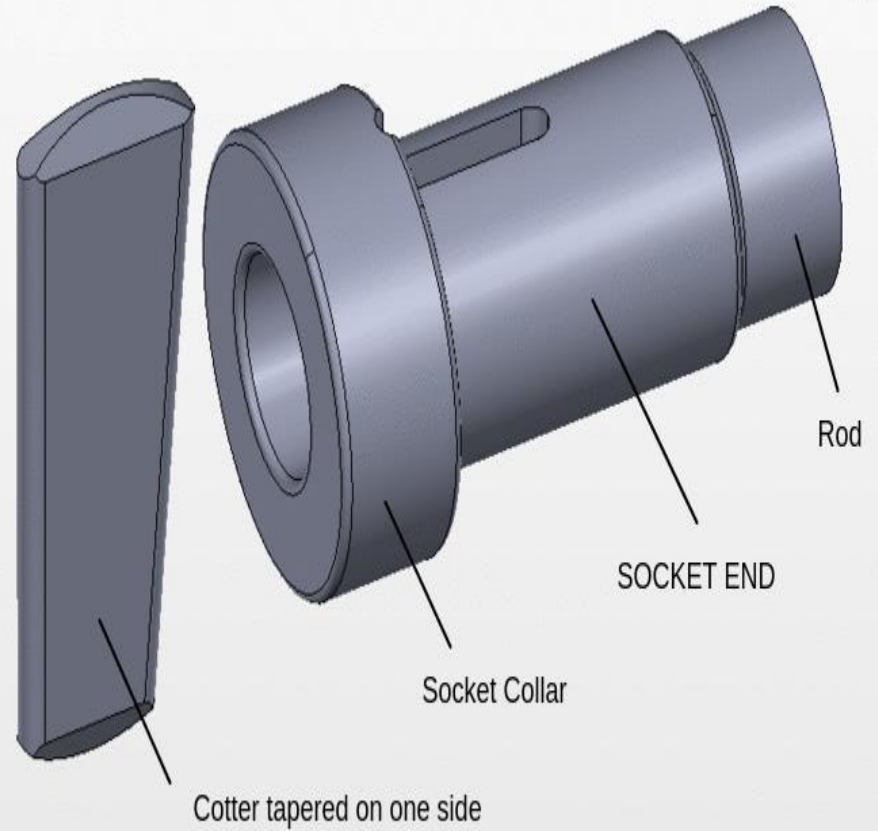
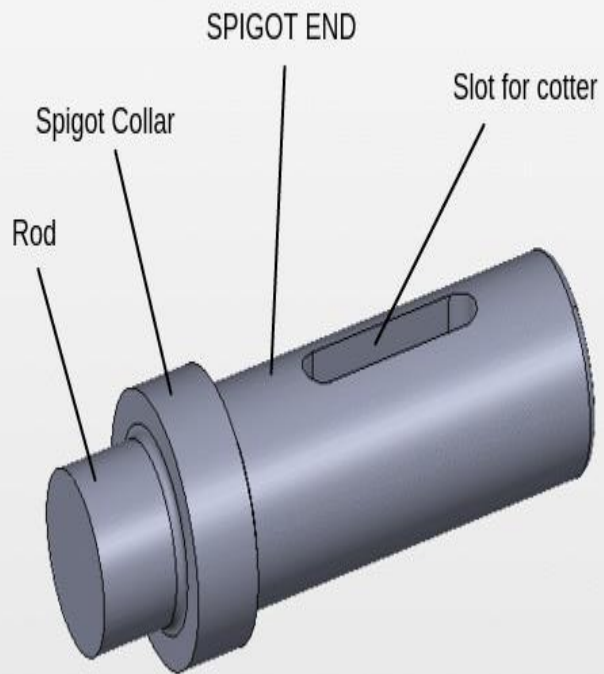
$$= \frac{\pi}{16} \times \tau \times d^3 \times e = \frac{\pi}{16} \times 56 \times (40)^3 \times 0.8125 = 571\,844\text{ N}$$

shear strength of the key

$$= l \times w \times \tau \times \frac{d}{2} = 75 \times 10 \times 56 \times \frac{40}{2} = 840\,000\text{ N}$$

$$\therefore \frac{\text{Shear strength of the key}}{\text{Normal strength of the shaft}} = \frac{840\,000}{571\,844} = 1.47$$

Cotter Joints



INTRODUCTION

- A cotter is a flat wedge shaped piece of rectangular cross-section and its width is tapered (either on one side or both sides) from one end to another for an easy adjustment.
- The taper varies from 1 in 48 to 1 in 24 and it may be increased up to 1 in 8, if a locking device is provided. The locking device may be a taper pin or a set screw used on the lower end of the cotter.
- The cotter is usually made of mild steel or wrought iron.
- A cotter joint is a temporary fastening and is used to connect rigidly two co-axial rods or bars which are subjected to axial tensile or compressive forces.
- It is usually used in connecting a piston rod to the crosshead of a reciprocating steam engine, a piston rod and its extension as a tailer pump rod, strap end of connecting rod etc.

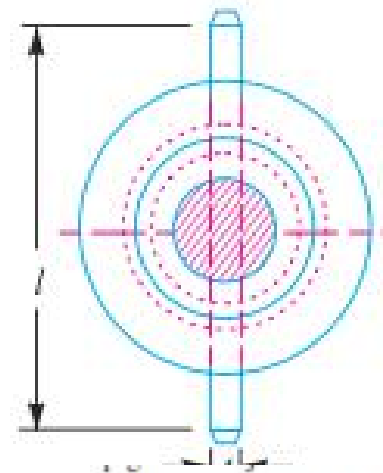
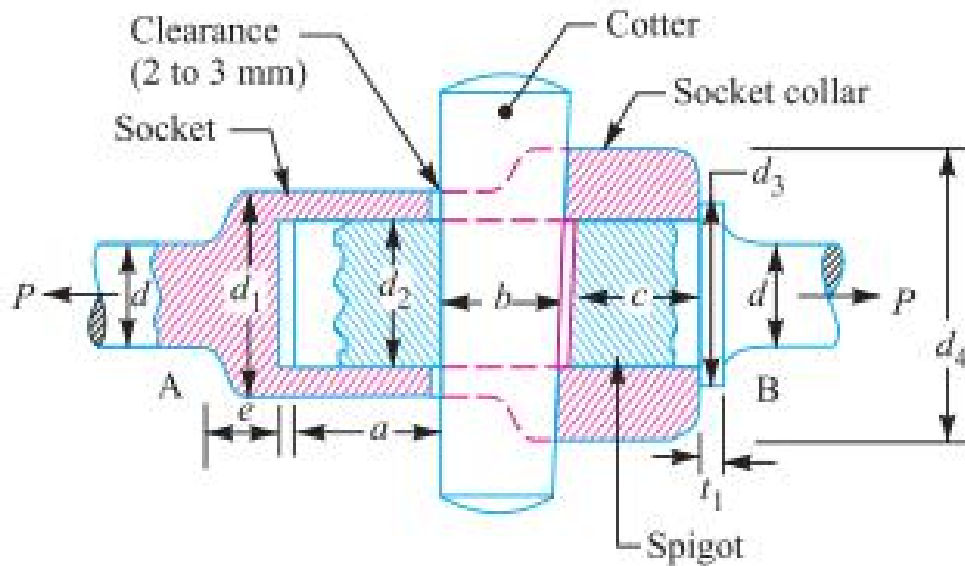
Types of Cotter Joints



Following are the three commonly used cotter joints to connect two rods by a cotter:

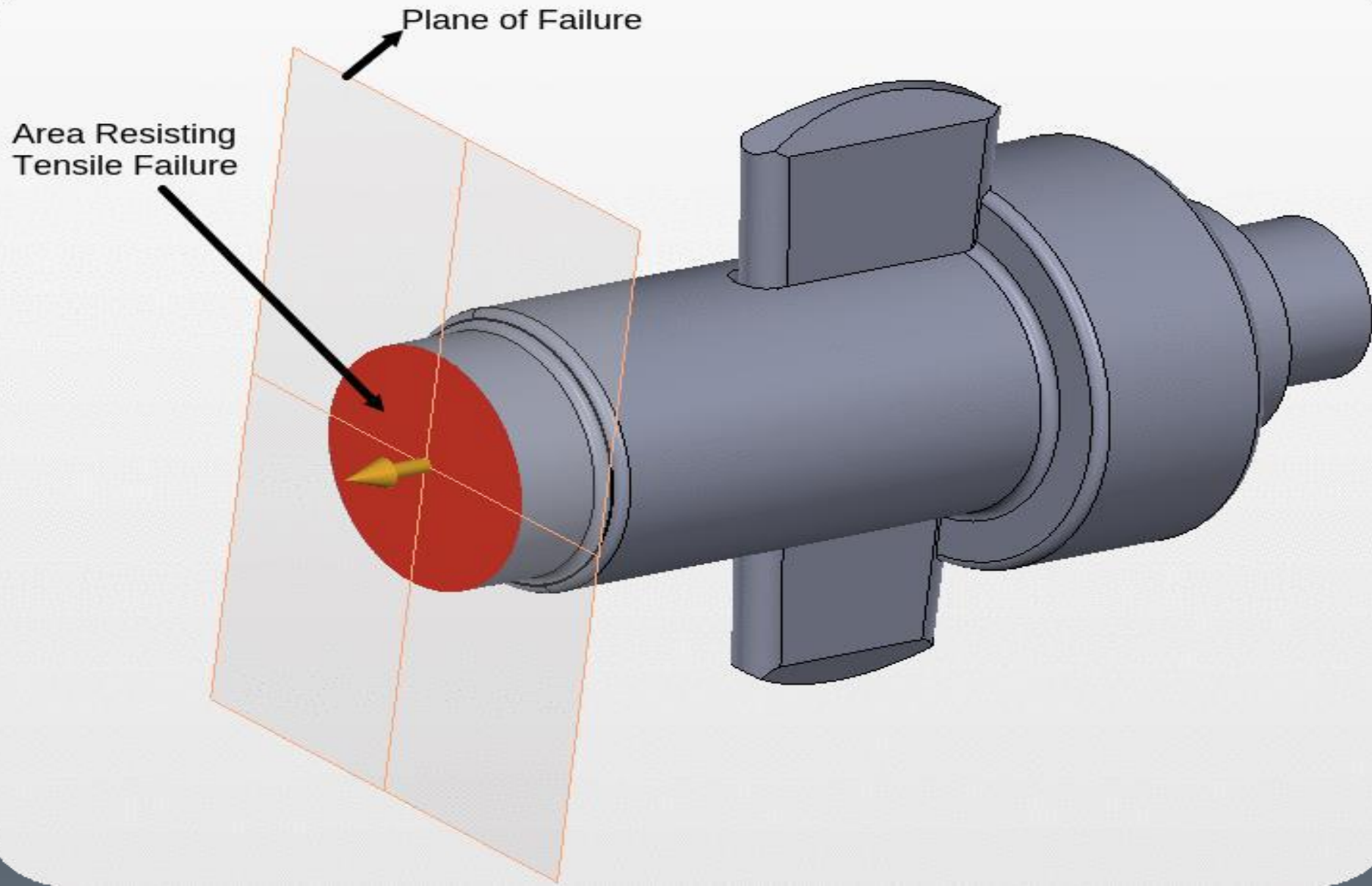
1. Socket and spigot cotter joint,
2. Sleeve and cotter joint, and
3. Gib and cotter joint

Socket and spigot Cotter joint



- P = Load carried by the rods,
- d = Diameter of the rods,
- d_1 = Outside diameter of socket,
- d_2 = Diameter of spigot or inside diameter of socket,
- d_3 = Outside diameter of spigot collar,
- t_1 = Thickness of spigot collar,
- d_4 = Diameter of socket collar,
- c = Thickness of socket collar,
- b = Mean width of cotter,
- t = Thickness of cotter,
- l = Length of cotter,
- a = Distance from the end of the slot to the end of rod,
- σ_t = Permissible tensile stress for the rods material,
- τ = Permissible shear stress for the cotter material, and
- σ_c = Permissible crushing stress for the cotter material.

1) Failure of the rod in tension



(Tensile failure of Rod).

The rod may fail in tension due to the tensile load P .

Area resisting tearing = $(\pi/4) * d^2$

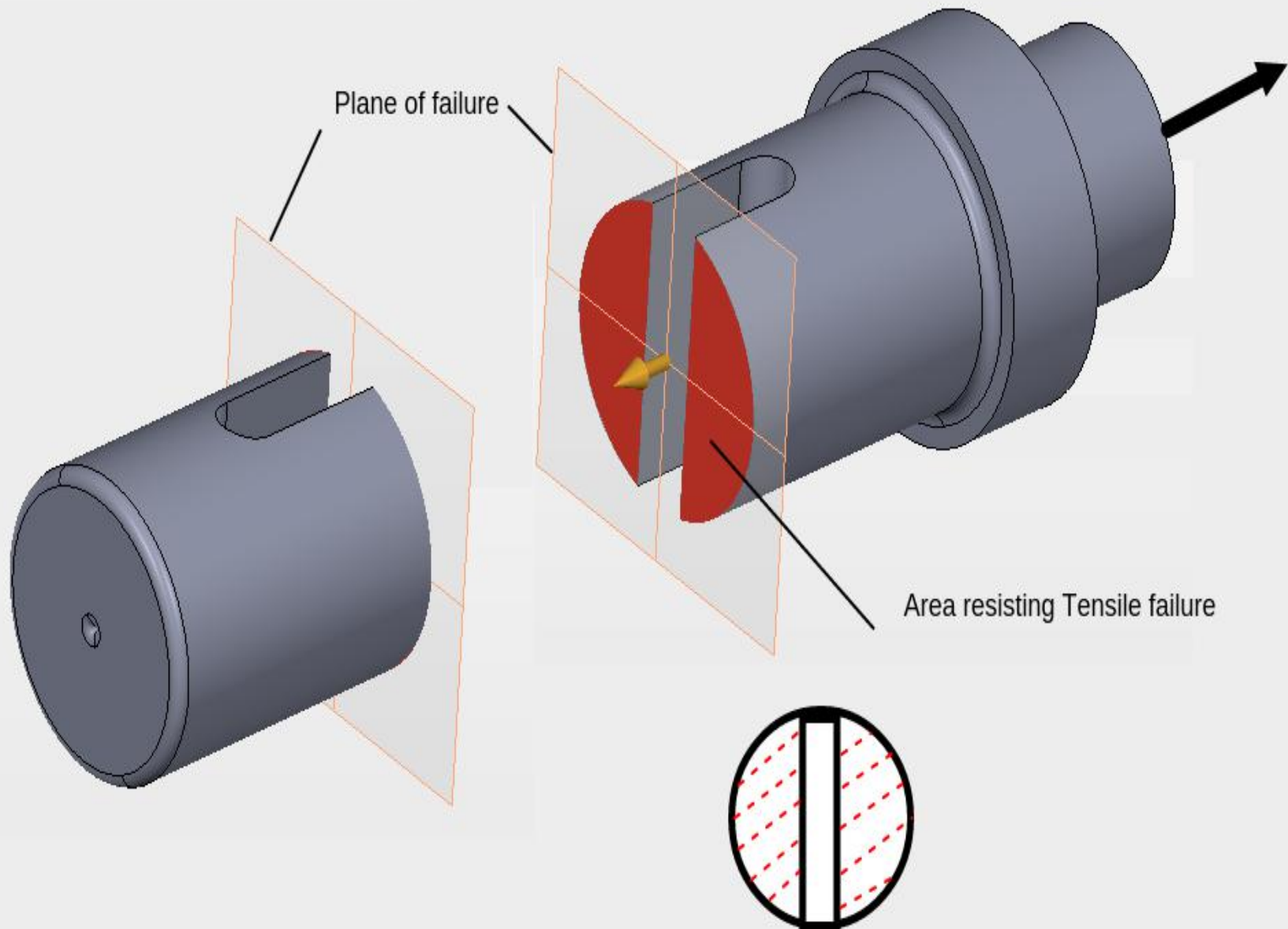
Tearing strength of rods = $(\pi/4) * (d^2) * (\sigma_t)$

Equating this to load (P), we have

$$P = [(\pi/4) * (d)^2] * \sigma_t$$

From this equation, diameter of the rod (d) may be calculated

2) Failure of the Spigot in Tension across the weakest section (or slot)



Cont....

Since the weakest section of the spigot is that section which has a slot in it for the cotter, therefore

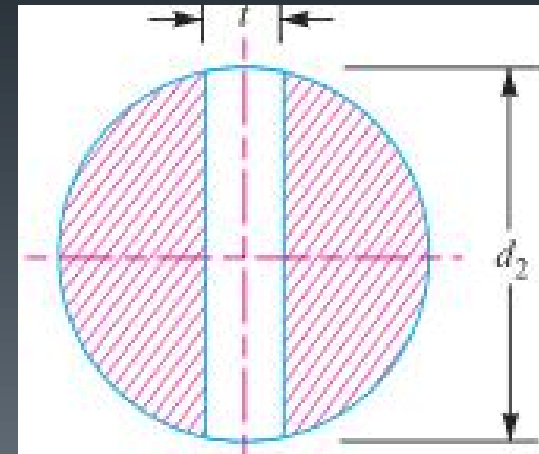
- Area resisting tearing $A_t = \frac{\pi}{4} d_2^2 - d_2 t$
- Tearing strength of the rods, $P_t = [\frac{\pi}{4} d_2^2 - d_2 t] \sigma_t$
- To avoid failure tearing strength \geq Applied load, $P_t \geq P$
- In limiting condition $P = P_t$

$$P = [\frac{\pi}{4} d_2^2 - d_2 t] \sigma_t$$

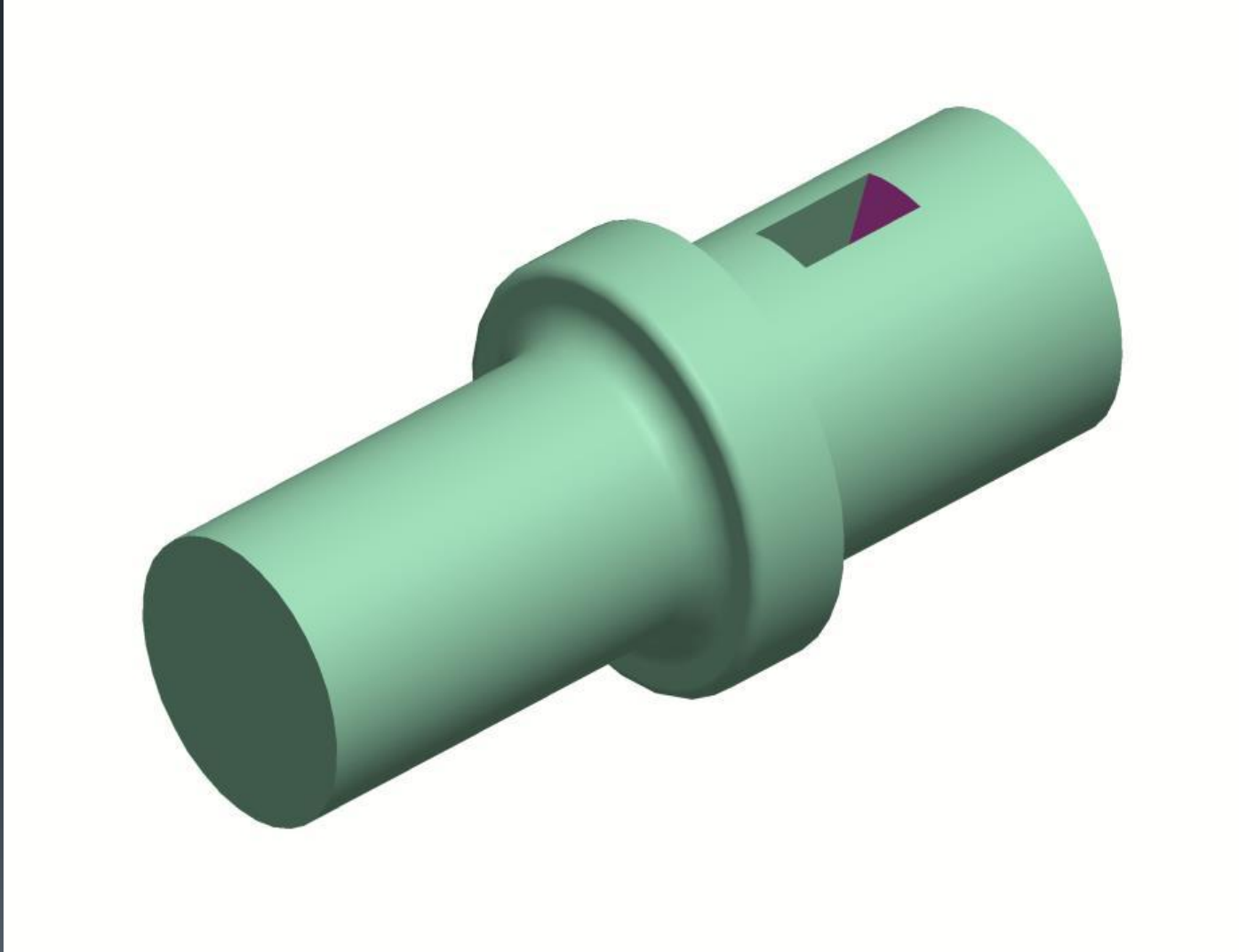
In above equation both d_2 & t are unknowns, take an assumption

$$t = d_2/4$$

d_2 may be calculated



3) Failure of the rod or cotter in crushing



Cont....

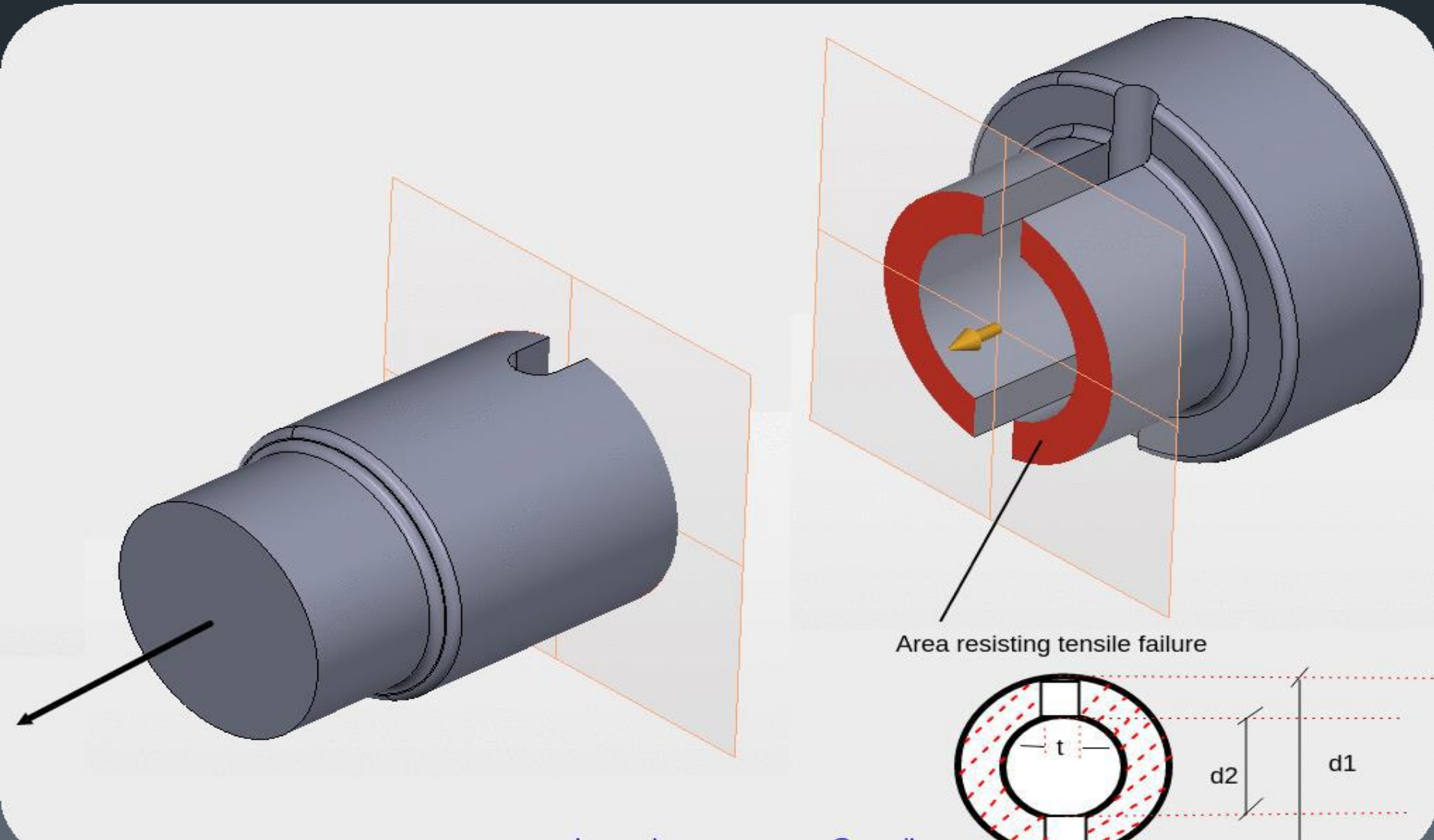
- Area that resists crushing of a rod or cotter $A_c = d_2 t$
- Crushing strength of the rods, $P_c = d_2 t \sigma_c$
- To avoid failure crushing strength \geq Applied load, $P_c \geq P$
- In limiting condition $P = P_c$

$$P = d_2 t \sigma_c$$

From above equation induced crushing strength may be verified, i.e., induced stress should be less than allowable stress.

Then finalise d_2 & t

4) Failure of the socket in tension across the slot

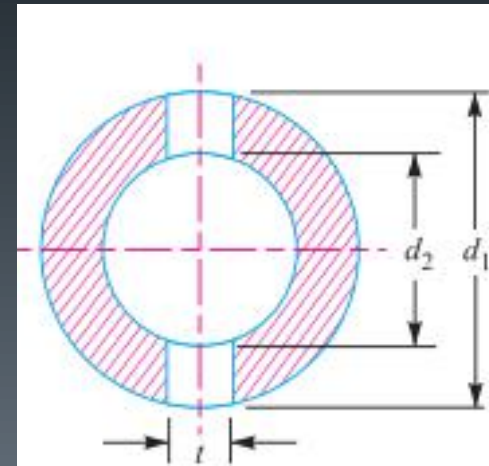


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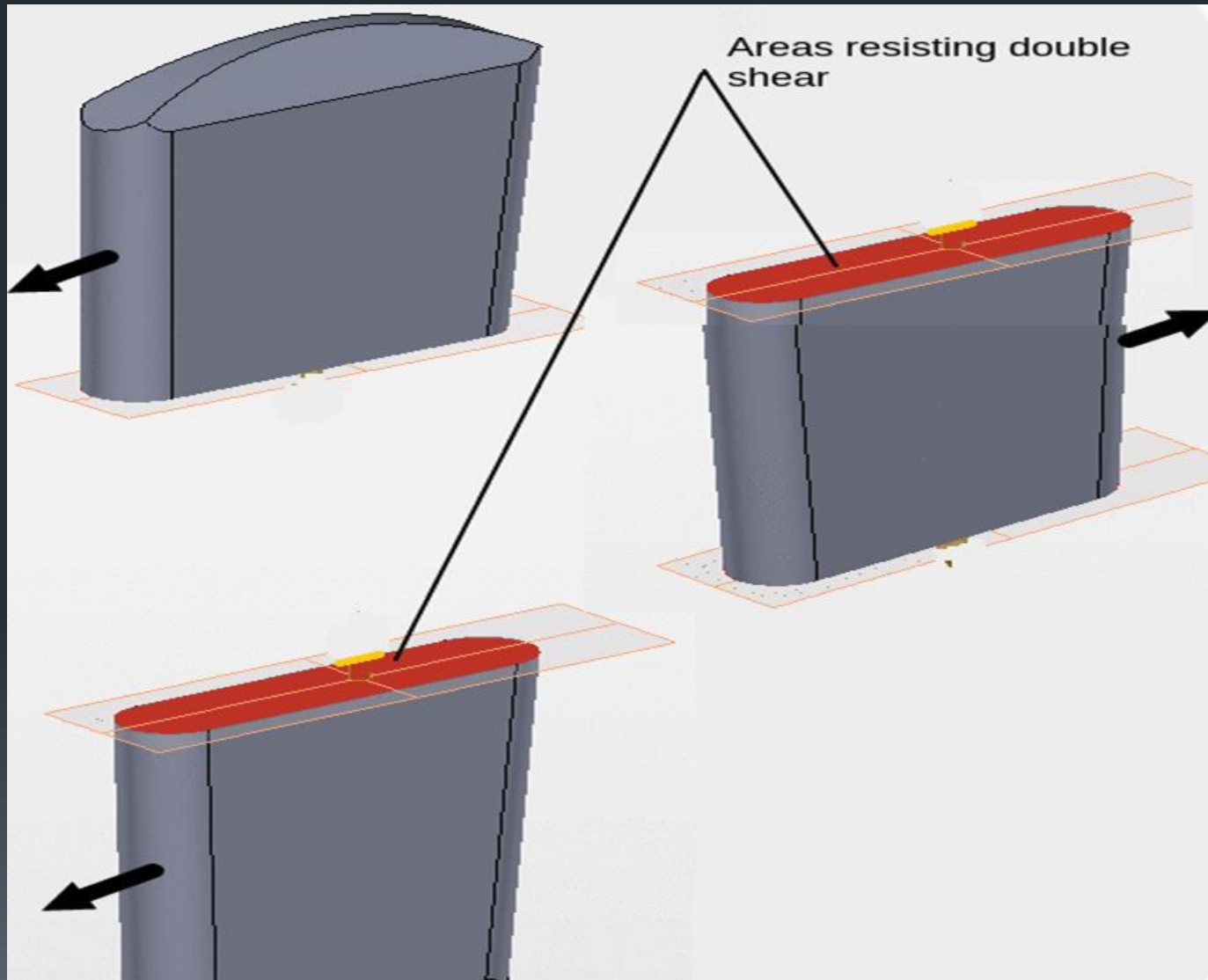
- Area resisting tearing $A_t = \frac{\pi}{4} [d_1^2 - d_2^2] - (d_1 - d_2)t$
- Tearing strength of the rods, $P_t = \left\{ \frac{\pi}{4} [d_1^2 - d_2^2] - (d_1 - d_2)t \right\} \sigma_t$
- To avoid failure tearing strength \geq Applied load
- $P_t \geq P$
- In limiting condition $P = P_t$

$$P = \left\{ \frac{\pi}{4} [d_1^2 - d_2^2] - (d_1 - d_2)t \right\} \sigma_t$$

From above equation outside diameter of socket may be calculated



5) Failure of the cotter in shear



Cont....

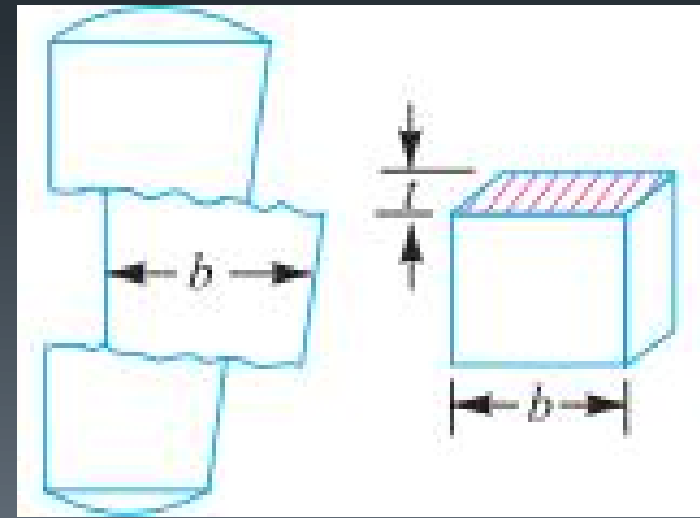
- Area resisting shearing $A_s = 2 b t$
- Shearing strength of the rods, $P_s = 2 b t \tau$
- To avoid failure shearing strength \geq Applied load

$$P_s \geq P$$

- In limiting condition $P = P_s$

$$P = 2 b t \tau$$

From above equation mean width of cotter may be calculated



6) Failure of socket collar in crushing

Crushing Failure of Socket End or Socket Collar

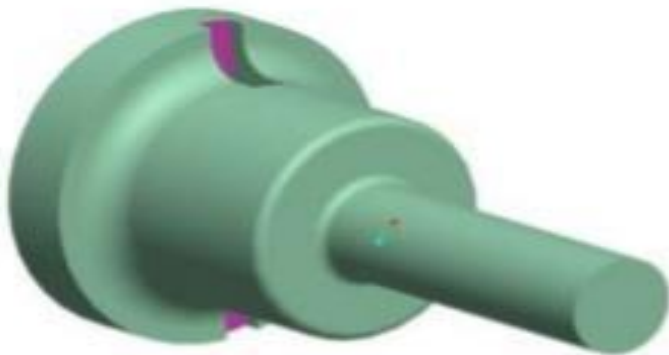
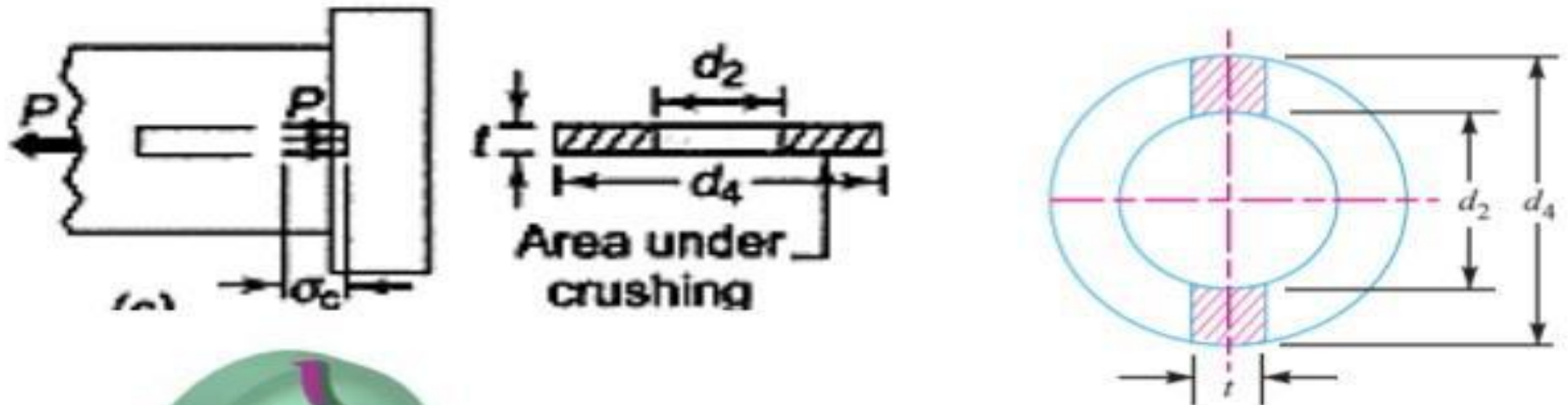


FIG 08: CRUSHING OF COTTER PIN AGAINST SOCKET

$$\sigma_c = \frac{P}{(d_4 - d_2) t}$$

From this Eq. diameter d_4 can be obtained.

Cont....

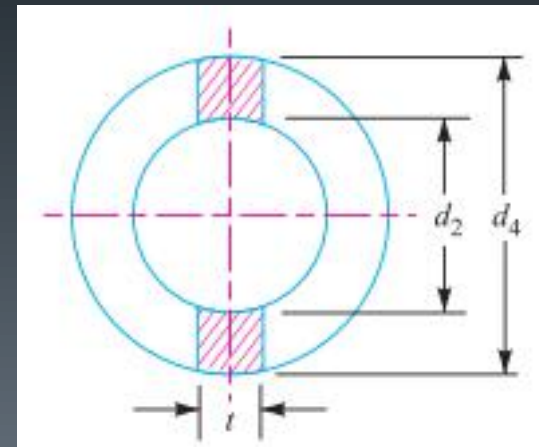
- Area resisting shearing $A_c = (d_4 - d_2) t$
- Crushing strength of the rods, $P_c = (d_4 - d_2) t \sigma_c$
- To avoid failure shearing strength \geq Applied load

$$P_c \geq P$$

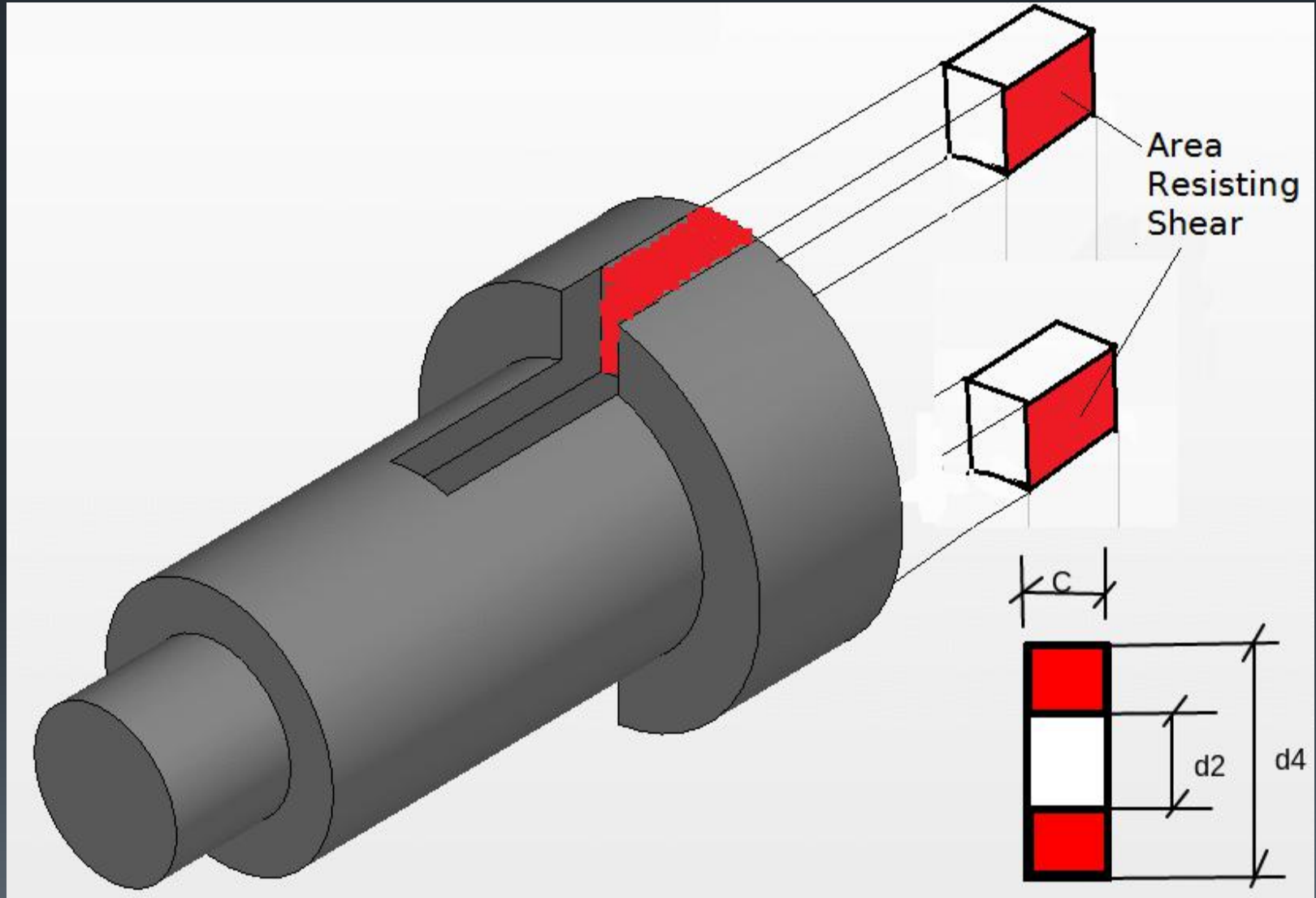
- In limiting condition $P = P_c$

$$P = (d_4 - d_2) t \sigma_c$$

From above equation diameter of socket collar may be calculated



7) Failure of socket end in shearing



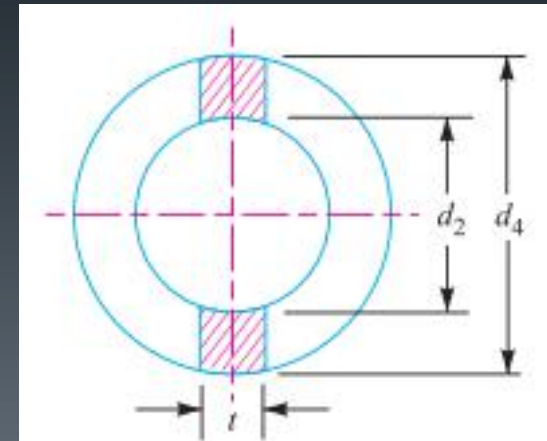
- Area resisting shearing $A_s = (d_4 - d_2) c$
- Shearing strength of the rods, $P_s = (d_4 - d_2) c \tau$
- To avoid failure shearing strength \geq Applied load

$$P_s \geq P$$

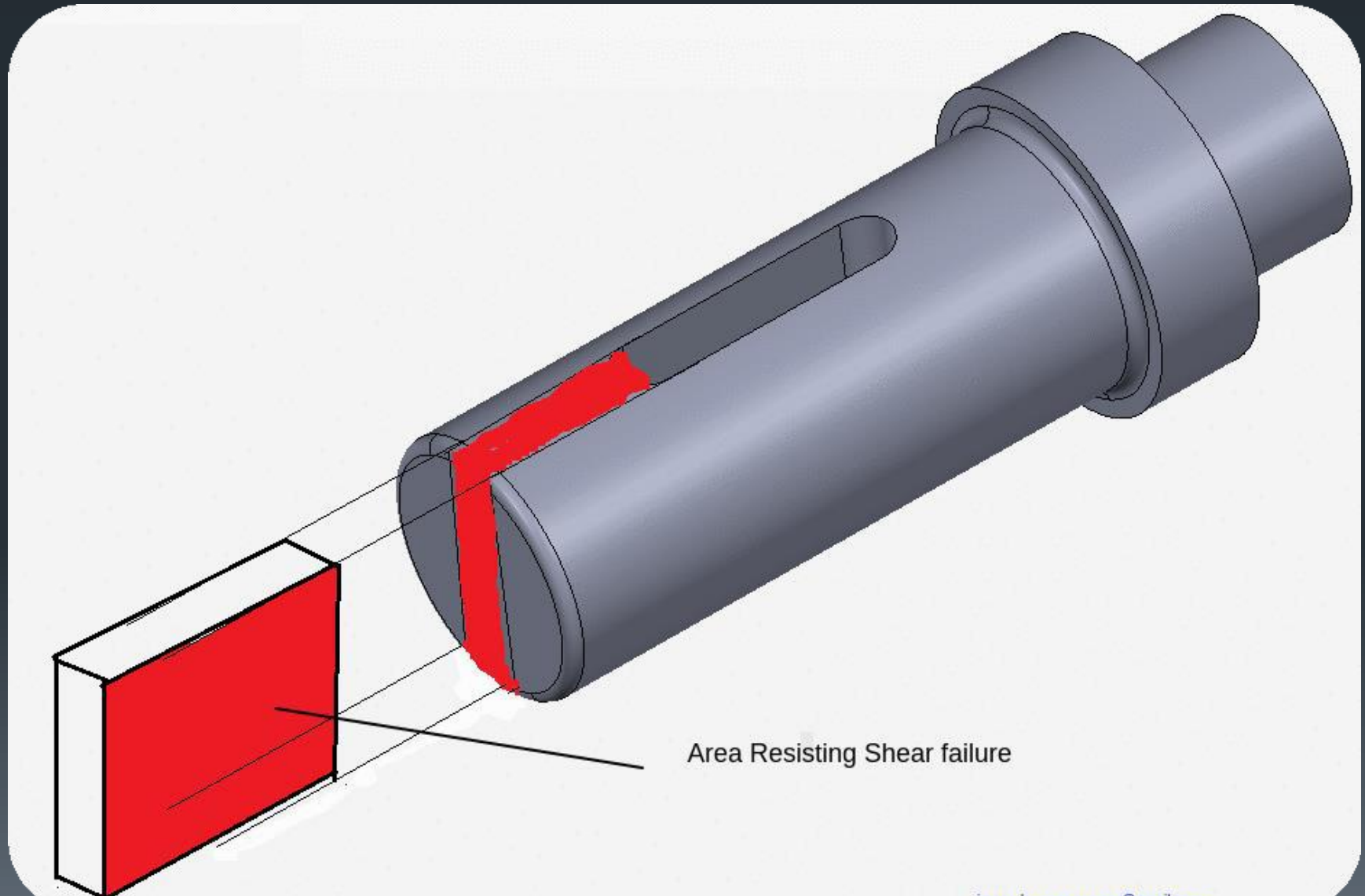
- In limiting condition $P = P_c$

$$P = (d_4 - d_2) c \tau$$

From above equation 'c' may be calculated



8) Failure of rod end in shear



- Area resisting shearing $A_s = 2 a d_2$
- Shearing strength of the rod end, $P_s = 2 a d_2 \tau$
- To avoid failure shearing strength \geq Applied load

$$P_s \geq P$$

- In limiting condition $P = P_c$

$$P = 2 a d_2 \tau$$

From above equation 'a' may be calculated

9) Failure of spigot collar in crushing

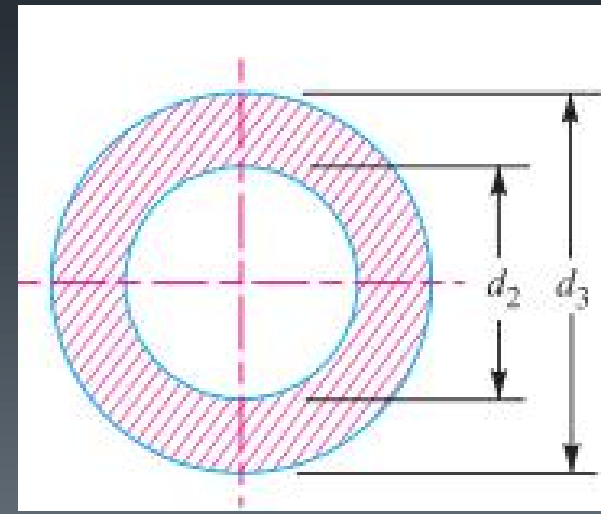
- Area resisting shearing $A_s = \frac{\pi}{4} (d_3^2 - d_2^2)$
- Crushing strength of the rods, $P_c = \frac{\pi}{4} (d_3^2 - d_2^2) \sigma_c$
- To avoid failure shearing strength \geq Applied load

$$P_c \geq P$$

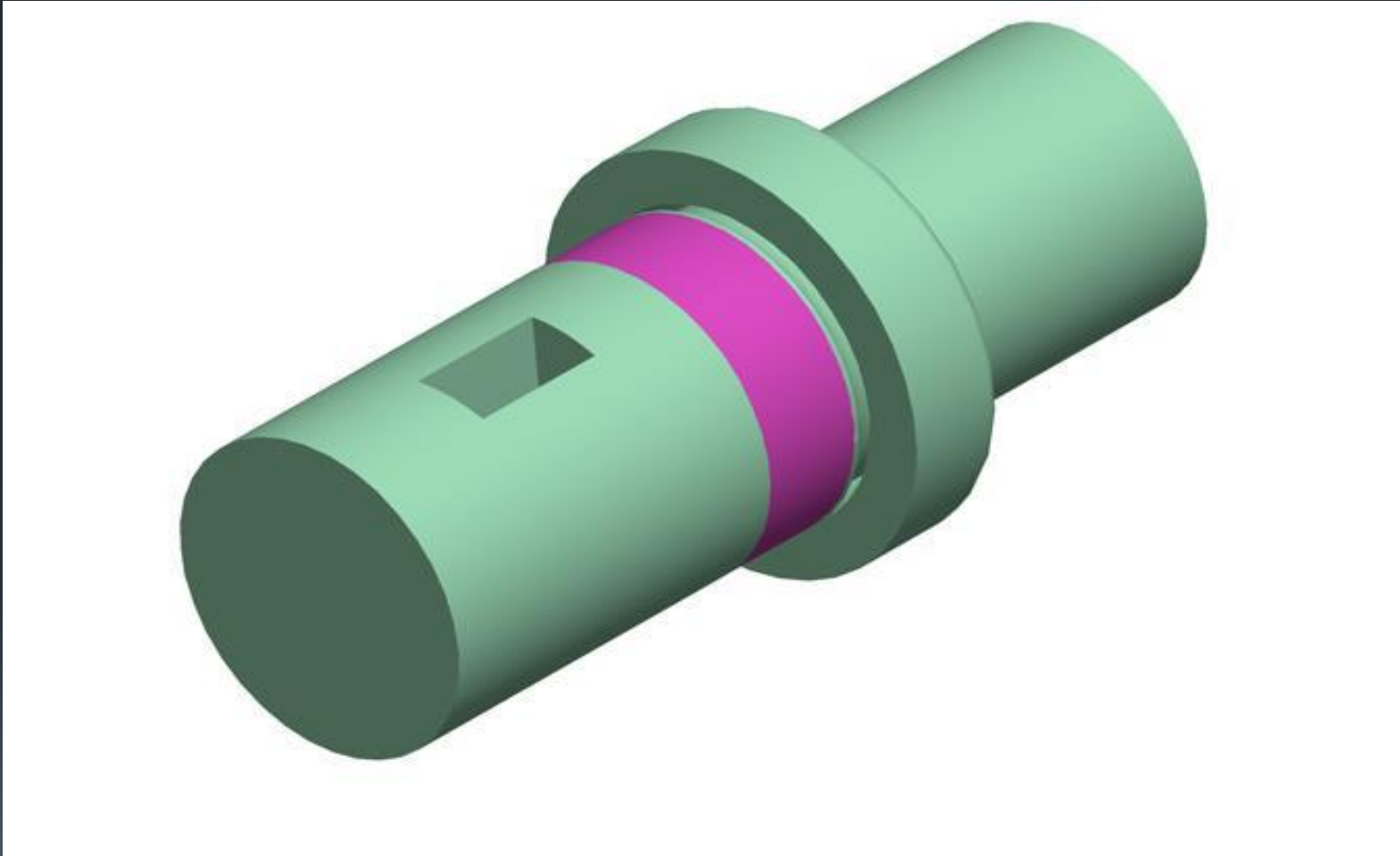
- In limiting condition $P = P_c$

$$P = \frac{\pi}{4} (d_3^2 - d_2^2) \sigma_c$$

From above equation diameter of spigot collar may be calculated



10) Failure of spigot collar in shearing



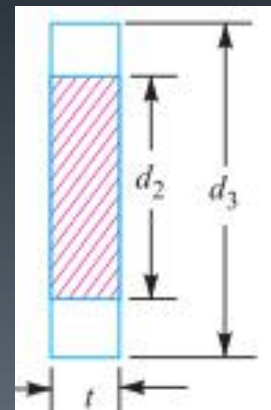
- Area resisting shearing $A_s = \pi d_2 t_1$
- Shearing strength of the rods, $P_s = \pi d_2 t_1 \tau$
- To avoid failure shearing strength \geq Applied load

$$P_s \geq P$$

- In limiting condition $P = P_s$

$$P = \pi d_2 t_1 \tau$$

From above equation thickness of spigot collar may be calculated



Failure of cotter in bending

- The maximum bending moment occurs at the centre of the cotter and is given by

$$M_{MAX} = P/2 [1/3 * (d_4 - d_2)/2 + d_2/2] - P/2 [d_2/4]$$

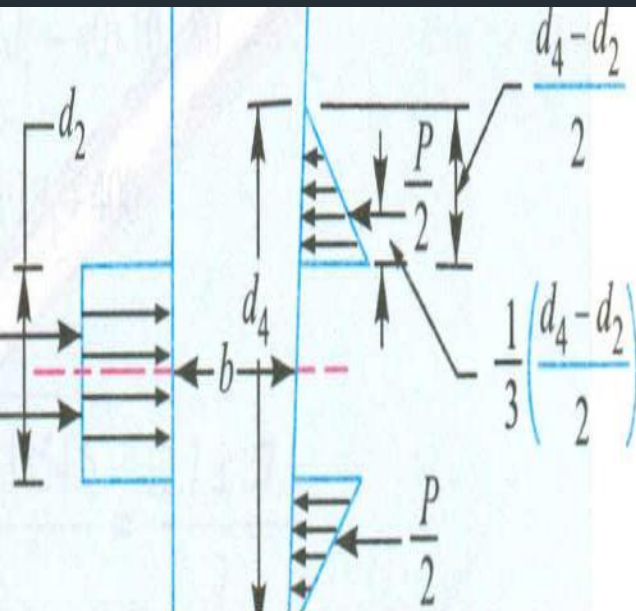
$$M_{MAX} = P/2 [(d_4 - d_2)/6 + d_2/2 - d_2/4]$$

$$M_{MAX} = P/2 [(d_4 - d_2)/6 + d_2/4]$$

We know that section modulus of the cotter ,
 $Z = t * b^2 / 6$

Bending stress induced in the cotter,

$$\sigma_b = (M_{MAX} / Z)$$



Failure of cotter in bending

Bending stress induced in the cotter,

$$\sigma_b = (M_{MAX}/Z)$$

Bending stress induced in the cotter,

$$\sigma_b = P/2[(d_4-d_2)/6+d_2/4]/(t*b^2/6)$$

Bending stress induced in the cotter,

$$\sigma_b = P[(d_4)+0.5 *d_2]/(2* t*b^2)$$

This bending stress induced in the cotter should be less than the allowable bending stress of the cotter.





Design of Couplings

Presented by

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Department of Mechanical Engineering

Unit 5 Design of Shafts and Couplings

Shafts: Design of solid and hollow shafts bending, torsion, axial and combined bending and axial loading.

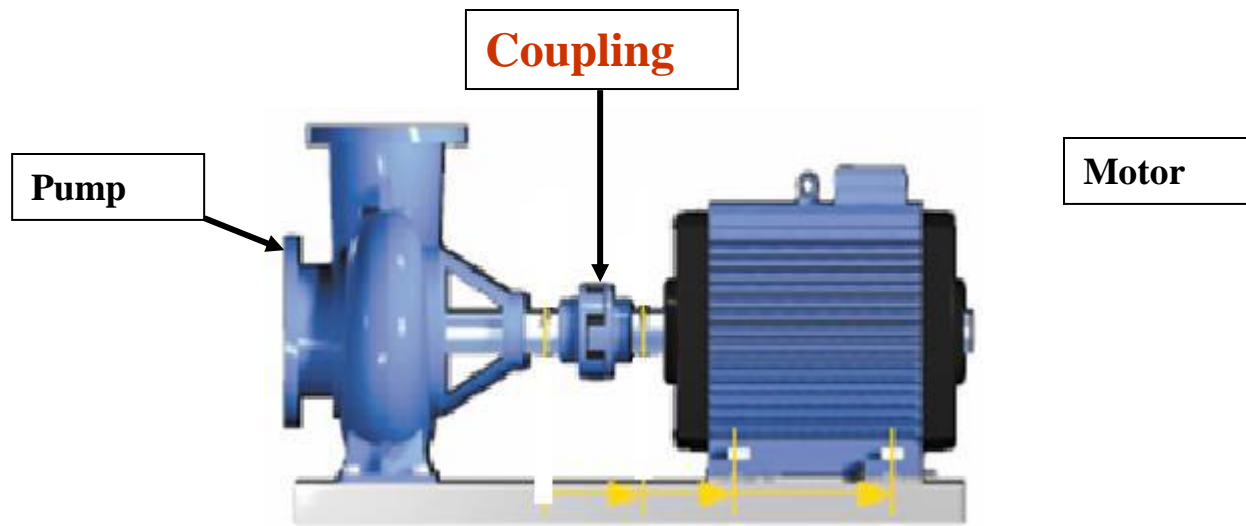
Shaft Couplings: Design of Rigid couplings-Muff, Split muff and Flange couplings-Flexible couplings: bushed pin type.

Course Outcome 5

At the end of the topic, student will be able to **Design of shaft couplings for given conditions.**

Couplings

Coupling is a device used to connect two shafts together at their ends for the purpose of transmitting power



Uses of coupling

- To provide connection of shafts of units made separately
- To allow misalignment of the shafts or to introduce mechanical flexibility.
- To reduce the transmission of shock loads
- To introduce protection against overloads.
- To alter the vibration characteristics

Types of Couplings

1) Rigid Couplings

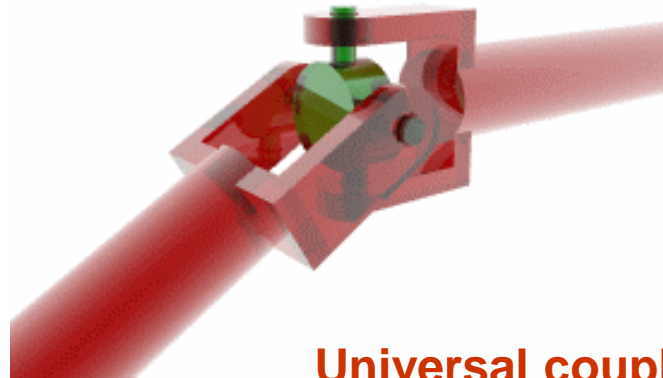


Rigid coupling

2) Flexible Couplings



Flexible coupling



Universal coupling

Rigid Couplings

- It is used to connect two shafts which are perfectly aligned.

Types of Rigid Couplings are

- 1) Sleeve or Muff coupling
- 2) Clamp or Split-Muff coupling
- 3) Flange Coupling

Flexible Couplings

- It is used to connect two shafts having both lateral and angular misalignment.

Types of Flexible Couplings are

- 1) Bushed pin type coupling,
- 2) Universal coupling, and
- 3) Oldham Coupling

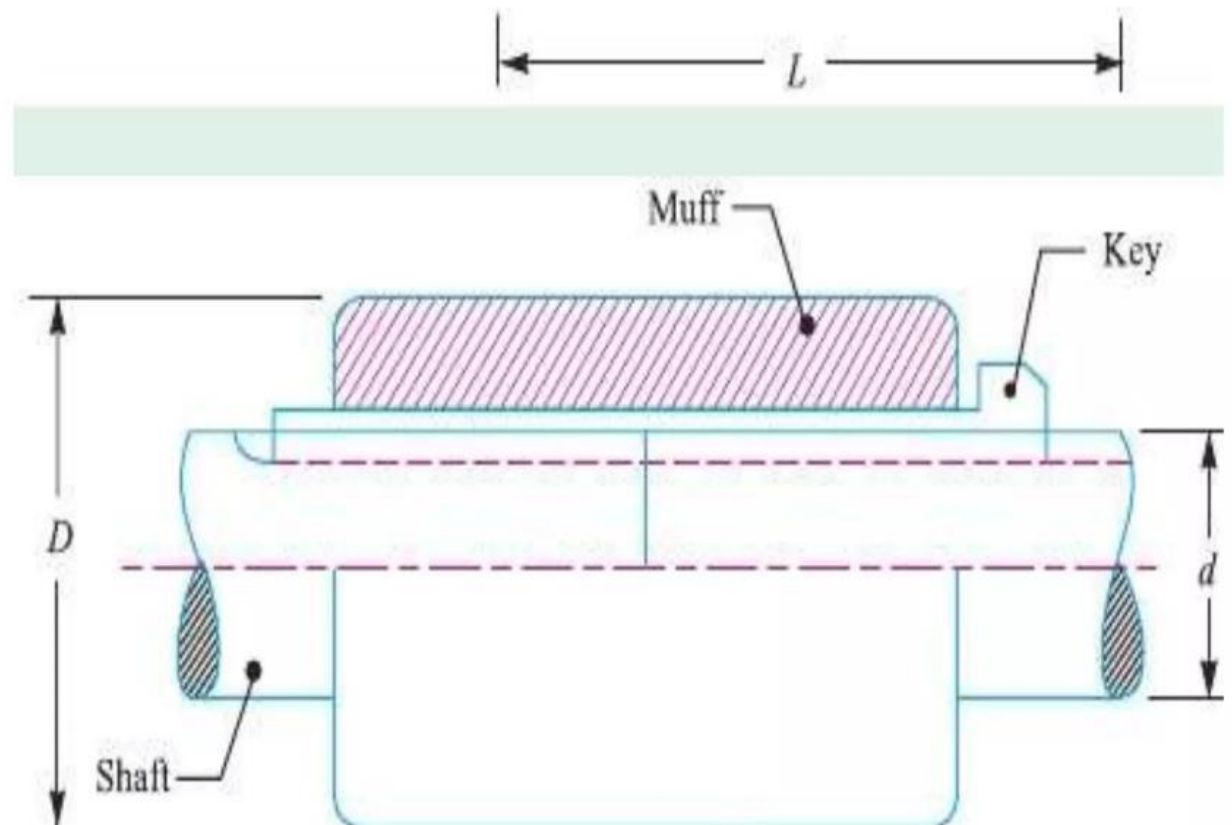
1. Sleeve or Muff-Coupling

- It is the simplest type, made of cast iron.
- It consists of sleeve and is fitted over the ends of the two shafts by means of gib head key
- Power is transmitted by means of a key and a sleeve

Design of Muff Coupling

The following parts are to be designed

- Shaft
- Sleeve
- Key



1) Design of Shaft

- Based on the loading the design of shaft is to be done as per earlier discussion
- Diameter of the shaft (d) is obtained

2) Design of Sleeve

Let T = Torque to be transmitted by the coupling, and

τ_c = Permissible shear stress for the material of the sleeve which is cast iron = 14 Mpa

$$T = \frac{\pi}{16} \times \tau_c \left(\frac{D^4 - d^4}{D} \right) = \frac{\pi}{16} \times \tau_c \times D^3 (1 - k^4)$$

3) Design of Key

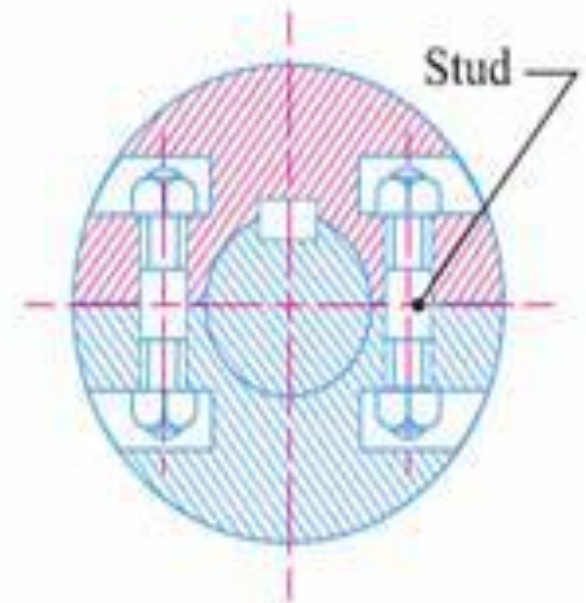
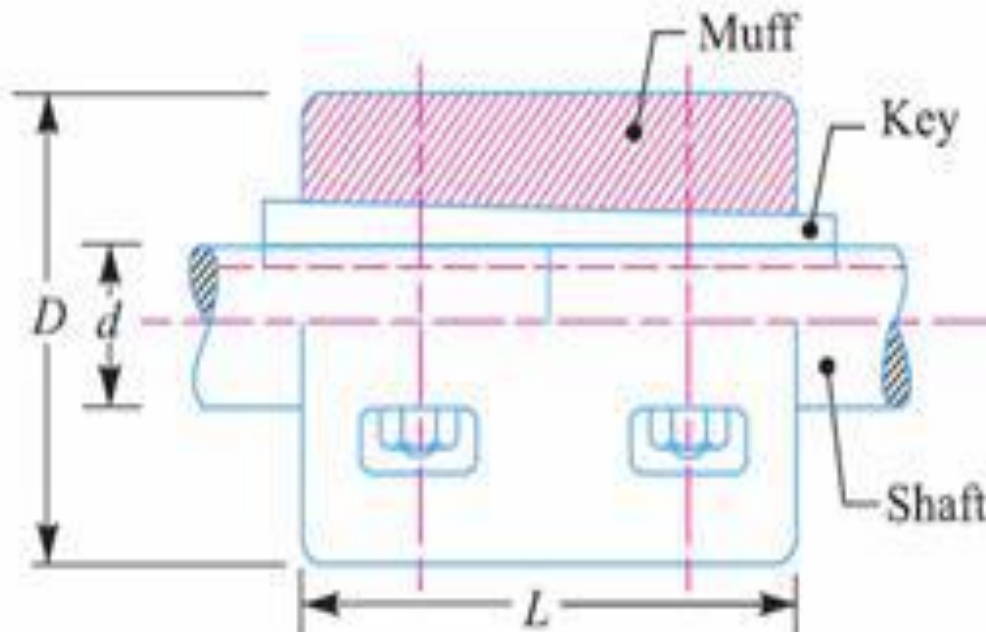
- Length of the key is $l=L/2 = 3.5d/2$
- After fixing the length of the Key the in each shaft, the induced shearing and crushing stresses may be checked. We know that torque transmitted,

$$T = F \times \frac{d}{2} = l \times \frac{t}{2} \times \sigma_c \times \frac{d}{2}$$

$$T = F \times \frac{d}{2} = l \times w \times \tau \times \frac{d}{2}$$

Design of Split Muff Coupling

- In this coupling, sleeve or muff is made in two halves, which are split along the plane passing through the axes of the shafts and are bolted together .
- The halves of the muff are made of cast iron.



Cont....

- The shaft ends are made to a butt each other
- A single key is fitted directly in the keyways of both the shafts.
- One-half of the muff is fixed from below and the other half is placed from above.
- Both the halves are held together by means of mild steel studs or bolts and nuts.
- The number of bolts may be two, four or six. The nuts are recessed into the bodies of the muff castings.

Applications: This coupling may be used for heavy duty and moderate speeds.

a) Design of Muff and Key

- The usual proportions of the muff for the clamp or compression coupling are :

Diameter of the muff or sleeve, $D = 2d + 13 \text{ mm}$

Length of the muff or sleeve, $L = 3.5 d$

where $d =$ Diameter of the shaft.

Let $T =$ Torque to be transmitted by the coupling, and

$\tau_c =$ Permissible shear stress for the material of the sleeve
which is cast iron = 14 Mpa

$$T = \frac{\pi}{16} \times \tau_c \left(\frac{D^4 - d^4}{D} \right) = \frac{\pi}{16} \times \tau_c \times D^3 (1 - k^4)$$

- Length of the key is $l=L/2 = 3.5d/2$
- After fixing the length of the Key the in each shaft, the induced shearing and crushing stresses may be checked. We know that torque transmitted,

$$T = F \times \frac{d}{2} = l \times \frac{t}{2} \times \sigma_c \times \frac{d}{2}$$

$$T = F \times \frac{d}{2} = l \times w \times \tau \times \frac{d}{2}$$

b) Design of Clamp

- The force exerted by each bolt, $F = (\pi/4) (d_b^2) \sigma_t$
- Force exerted by the bolts on each side of the shaft

$$F = (\pi/4) (d_b^2) \sigma_t \times (n/2)$$

- Let p be the pressure on the shaft and the muff surface due to the force, then for uniform pressure distribution over the surface,

$$p = \frac{\text{Force}}{\text{Projected area}} = \frac{\frac{\pi}{4} (d_b)^2 \sigma_t \times \frac{n}{2}}{\frac{1}{2} L \times d}$$

Frictional force between each shaft and muff,

- $F = \mu \times \text{pressure} \times \text{area} = \mu \times p \times \frac{1}{2} \times \pi d \times L = \mu \times \frac{\pi^2}{8} \times (d_b^2) \sigma_t \times n$
- The torque that can be transmitted by the coupling,

$$T = F \times \frac{d}{2} = \mu \times \frac{\pi^2}{8} (d_b)^2 \sigma_t \times n \times \frac{d}{2} = \frac{\pi^2}{16} \times \mu (d_b)^2 \sigma_t \times n \times d$$

Flange Coupling

- A flange coupling usually applies to a coupling having two separate cast iron flanges.
- Each flange is mounted on the shaft end and keyed to it.
- The faces are turned up at right angle to the axis of the shaft

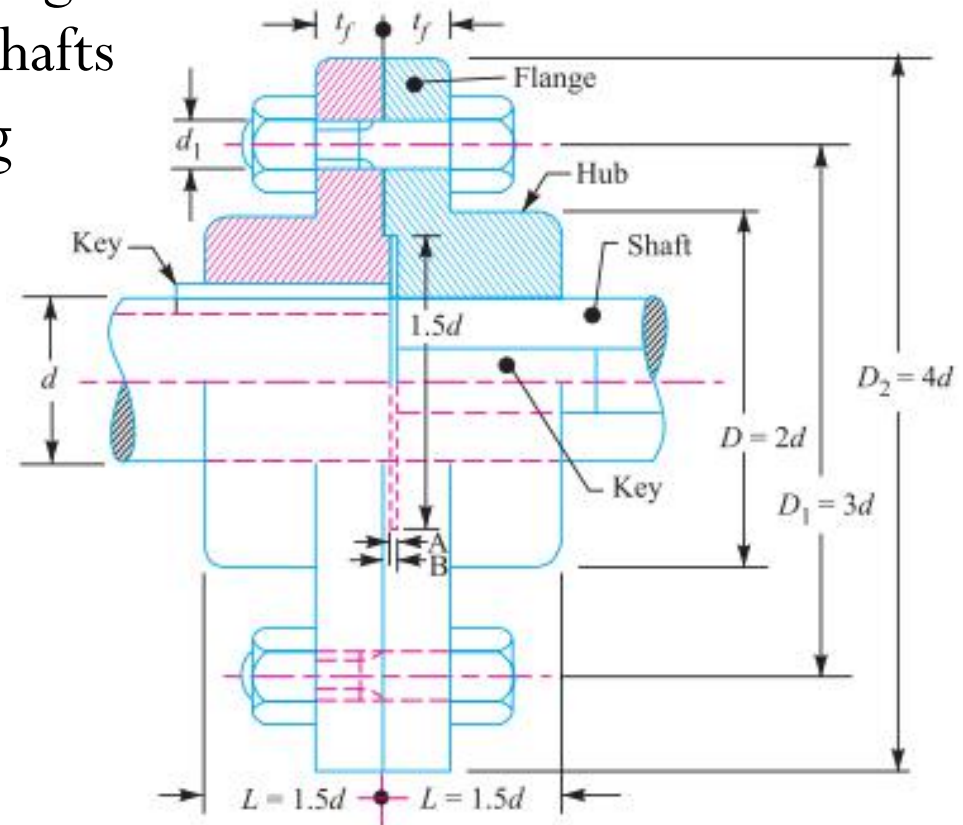
Flange coupling are

1. Unprotected type flange coupling
2. Protected type flange coupling
3. Marine type flange coupling



1. Unprotected type flange coupling

- Each shaft is keyed to the boss of a flange with a counter sunk key and the flanges are coupled together by means of bolts.
- Generally, three, four or six bolts are used.
- The keys are staggered at right angle along the circumference of the shafts in order to divide the weakening effect caused by keyways.



Cont...

- The usual proportions for an unprotected type cast iron flange couplings, are as follows :

If d is the diameter of the shaft or inner diameter of the hub, then

Outside diameter of hub, $D = 2 d$

Length of hub, $L = 1.5 d$

Pitch circle diameter of bolts, $D_1 = 3d$

Outside diameter of flange, $D_2 = D_1 + (D_1 - D) = 2 D_1 - D = 4 d$

Thickness of flange, $t_f = 0.5 d$

Number of bolts = 3, for d upto 40 mm

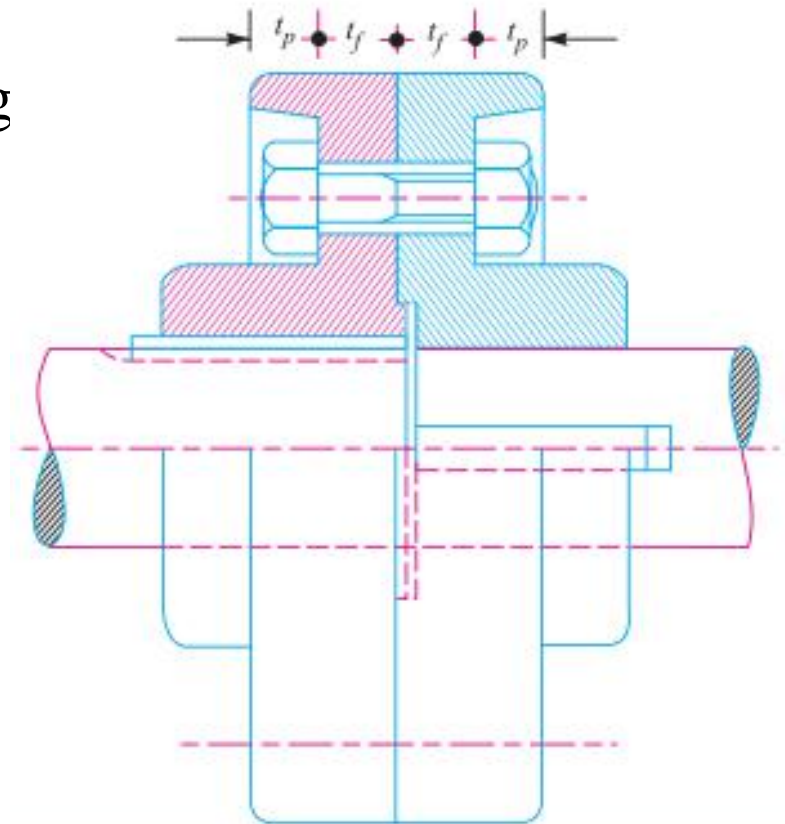
= 4, for d upto 100 mm

= 6, for d upto 180 mm

2. Protected type flange coupling.

- In a protected type flange coupling, as shown in Fig., the protruding bolts and nuts are protected by flanges on the two halves of the coupling, in order to avoid danger to the workman.
- The thickness of the protective circumferential flange (t_p) is taken as $0.25 d$.

The other proportions of the coupling are same as for unprotected type flange coupling.



3. Marine type flange coupling.

- In a marine type flange coupling, the flanges are forged integral with the shafts as shown in Fig. The flanges are held together by means of tapered headless bolts, numbering from four to twelve depending upon the diameter of shaft.
- The other proportions for the marine

type flange coupling are taken as follows :

Thickness of flange = $d / 3$

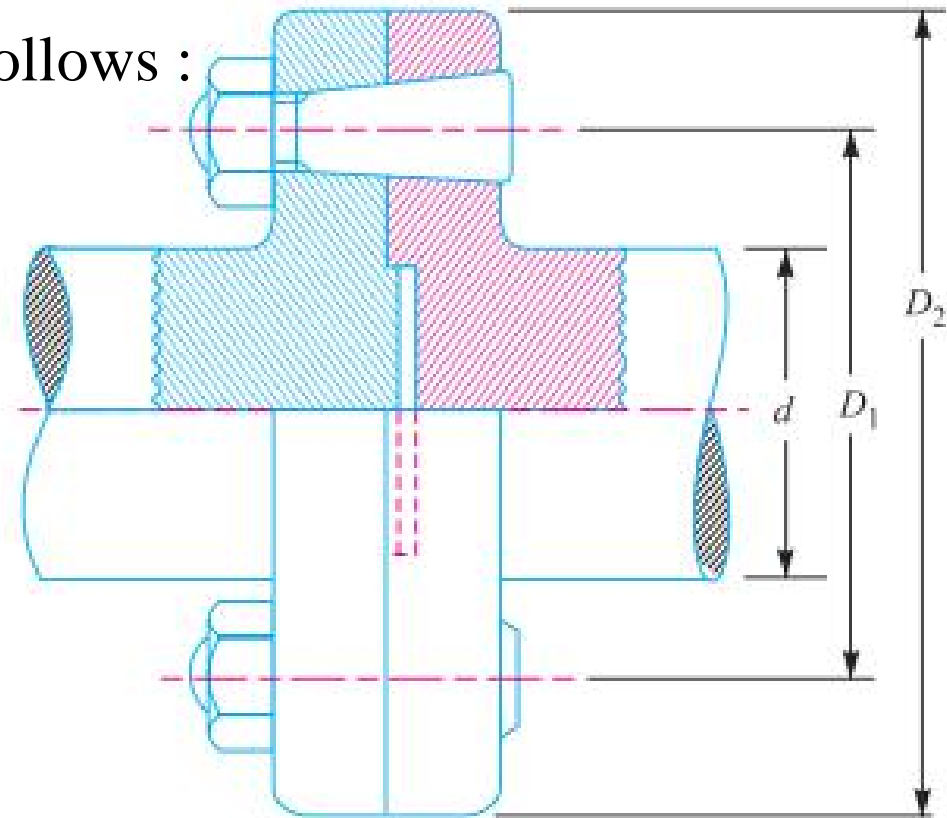
Taper of bolt = 1 in 20 to 1 in 40

Pitch circle diameter of bolts,

$D_1 = 1.6 d$

Outside diameter of flange,

$D_2 = 2.2 d$



Design of Flange Coupling

Let d = Diameter of shaft or inner diameter of hub,

D = Outer diameter of hub,

d = Nominal or outside diameter of bolt,

D_1 = Diameter of bolt circle,

n = Number of bolts,

t_f = Thickness of flange,

τ_s , τ_b , and τ_k = Allowable shear stress for shaft, bolt and key material respectively

τ_c = Allowable shear stress for the flange material i.e. cast iron,

σ_{cb} , σ_{ck} = Allowable crushing stress for bolt and key material respectively.

The flange coupling is designed as discussed below :

1) Design of Hub

The hub is designed by considering it as a hollow shaft, transmitting the same torque (T) as that of a solid shaft.

$$T = \frac{\pi}{16} \times \tau_c \left(\frac{D^4 - d^4}{D} \right)$$

The outer diameter of hub is usually taken as twice the diameter of shaft. Therefore from the above relation, the induced shearing stress in the hub may be checked.

The length of hub (L) is taken as 1.5 d.

2) Design for Key

$$T = F \times \frac{d}{2} = l \times \frac{t}{2} \times \sigma_c \times \frac{d}{2}$$

$$T = F \times \frac{d}{2} = l \times w \times \tau \times \frac{d}{2}$$

3) Design for flange

The flange at the junction of the hub is under shear while transmitting the torque. Therefore, the torque transmitted,

$T =$ Circumference of hub \times Thickness of flange \times Shear stress of flange \times Radius of hub

$$= \pi \times D \times t_f \times \tau_c \times \frac{D}{2} = \frac{\pi D^2}{2} \times t_f \times \tau_c$$

4) Design for Bolts

- The bolts are subjected to shear stress due to the torque transmitted. The number of bolts (n) depends upon the diameter of shaft and the pitch circle diameter of bolts (D) is taken as $3 d$. We know that

$$\text{Load on each bolt} = \frac{\pi}{4} \times d_1^2 \times \tau_b$$

$$\text{Total load on all the bolts} = \frac{\pi}{4} \times d_1^2 \times \tau_b \times n$$

$$\text{and torque transmitted } T = \frac{\pi}{4} \times d_1^2 \times \tau_b \times n \times \frac{D_1}{2}$$

From this equation, the diameter of bolt (d_1) may be obtained. Now the diameter of bolt may be checked in crushing.

$$\text{Torque, } \mathbf{T = n \times d_1 \times t_f \times \sigma_{cb} \times \frac{D_1}{2}}$$

Flexible Coupling

- Flexible coupling is used to join the abutting ends of shafts which are not in exact alignment.
- Permits an axial misalignment of the shaft without undue absorption of the power which the shaft are transmitting.

Types of Flexible Coupling:

- 1) Bushed -pin type flexible coupling
- 2) Oldham's Coupling
- 3) Universal Coupling

1) Bushed –pin type flexible coupling

- A bushed-pin flexible coupling, as shown in Fig, is a modification of the rigid type of flange coupling.
- The coupling bolts are known as pins.
- The rubber or leather bushes are used over the pins.
- The two halves of the coupling are dissimilar in construction.
- A clearance of 5 mm is left between the face of the two halves of the coupling.
- There is no rigid connection between them and the drive takes place through the medium of the compressible rubber or leather bushes.

Cont..

- Bearing load acting on each pin,

$$W = p_b \times d_2 \times l$$

Total bearing load on the bush or pins $T = W \times n = p_b \times d_2 \times l \times n$

The torque transmitted by the coupling,

$$T = W \times n \times \frac{D_1}{2} = p_b \times d_2 \times l \times n \times \frac{D_1}{2}$$

Direct shear stress due to pure torsion in the coupling halves, $\tau = \frac{W}{\frac{\pi}{4} \times d_1^2}$

Assuming a uniform distribution of the load W along the bush, the maximum bending moment on the pin,

$$M = W \left(\frac{l}{2} + 5\text{mm} \right)$$

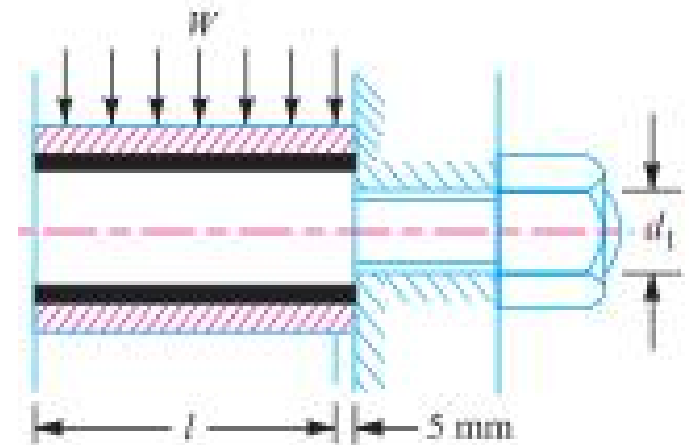
Bending stress, $\sigma_b = \frac{M}{Z} = \frac{W \left(\frac{l}{2} + 5\text{mm} \right)}{\frac{\pi}{4} \times d_1^3}$

Maximum principal stress

$$= \frac{1}{2} \left[\sigma + \sqrt{\sigma^2 + 4\tau^2} \right]$$

the maximum shear stress on the pin

$$= \frac{1}{2} \sqrt{\sigma^2 + 4\tau^2}$$



The value of maximum principal stress varies from 28 to 42 MPa.

Thank you